

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

TECHNICAL NOTE

No. 1673

TABLES AND CHARTS OF FLOW PARAMETERS

ACROSS OBLIQUE SHOCKS

By Mary M. Neice

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Langley Field, Va.



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SUMMARY

The oblique shock-wave equations have been solved for a range of Mach number in front of the shock from 1.05 to 4.0 and for a range of shock angle from a simple Mach wave to a normal shock. The results of these calculations are presented in tabular form and include values for the Mach number in front of the shock, the shock angle, the deviation of the flow across the shock, the Mach number behind the shock, the ratio of pressure behind the shock to pressure in front of the shock, the ratio of density in front of the shock to density behind the shock, and the change in entropy across the shock. Charts of several of these parameters are also presented.

INTRODUCTION

When a compression shock occurs in supersonic flow, the flow across the shock wave undergoes changes in its physical characteristics. These changes in characteristics of the flow behind the shock have been calculated in terms of similar characteristics of the flow in front of the shock from the fundamental shock-wave equations (reference 1). Calculations of values of some of the parameters most frequently used in supersonic studies have been presented in references 2 and 3. These computations, however, are limited in scope. More recently, reference 4 has been published to give a general survey of the equations, tables, and charts that are continually being used in research on supersonic flows. The material presented herein supplements and extends the information of reference 4. Because the nature of these computations is basic and the existence of such data would serve to avoid repetition of the same calculations in the future, an attempt has been made to present the results in a form that has already proved extremely useful for rapid calculations in the study of supersonic-flow problems and in supersonic design work.

SYMBOLS

M	Mach number
p	static pressure
ρ	density
ϵ	angle of the shock
δ	angle of deviation of the flow behind the shock
β	difference between angle of shock and deviation of the flow ($\epsilon - \delta$)
ΔS	change in entropy across shock, feet ² per second ² per degree Fahrenheit
R	gas constant for air (1715 ft ² /sec ² /°F)
γ	ratio of specific heats (1.4 for air)

Subscripts

1	conditions in front of shock
2	conditions behind shock
m	maximum
s	sonic

SHOCK EQUATIONS

The flow conditions associated with the phenomena of shock in a supersonic stream are illustrated in figure 1. This diagram shows the angle of shock ϵ , the deviation of flow across the shock δ , and the angle of flow behind the shock with relation to the front of the shock β . The evaluation of the flow parameters following a shock are included herein for ready reference. The equations used for the present calculations were taken from reference 1 and are as follows:

The deviation of the flow across a shock wave is obtainable from the relation:

$$\frac{1}{\tan \delta} = \left(\frac{\gamma + 1}{2} \frac{M_1^2}{M_1^2 \sin^2 \epsilon - 1} - 1 \right) \tan \epsilon \quad (1)$$

The value of the Mach number behind the shock is determined by

$$\frac{\tan \epsilon}{\tan \beta} = \frac{2}{\gamma + 1} \left(\frac{1}{M_2^2 \sin^2 \beta} + \frac{\gamma - 1}{2} \right) \quad (2)$$

The ratio of the pressure behind the shock to the pressure in front of the shock is

$$\frac{p_2}{p_1} = \frac{2\gamma}{\gamma + 1} \left(M_1^2 \sin^2 \epsilon - \frac{\gamma - 1}{2\gamma} \right) \quad (3)$$

and the ratio of density in front of the shock to density behind the shock is

$$\frac{\rho_1}{\rho_2} = \frac{2}{\gamma + 1} \left(\frac{1}{M_1^2 \sin^2 \epsilon} + \frac{\gamma - 1}{2} \right) \quad (4)$$

The variation of entropy across the shock is given by the equation

$$\Delta S = \frac{R}{(\gamma - 1)} \left(\log_e \frac{p_2}{p_1} + \gamma \log_e \frac{\rho_1}{\rho_2} \right) \quad (5)$$

where

$$R = 1715 \text{ ft}^2/\text{sec}^2/\text{°F}$$

The equation for the shock angle which gives maximum possible deviation of the flow across the shock is

$$\sin^2 \epsilon_m = \frac{1}{\gamma M_1^2} \left[\frac{\gamma + 1}{4} M_1^2 - 1 + \sqrt{(\gamma + 1) \left(1 + \frac{\gamma - 1}{2} M_1^2 + \frac{\gamma + 1}{16} M_1^4 \right)} \right] \quad (6)$$

The value of the angle of the shock which gives sonic velocity behind the shock is obtainable from the equation

$$\sin^2 \epsilon_s = \frac{1}{\gamma M_1^2} \left[\frac{\gamma + 1}{4} M_1^2 - \frac{3 - \gamma}{4} + \sqrt{(\gamma + 1) \left(\frac{9 + \gamma}{16} - \frac{3 - \gamma}{8} M_1^2 + \frac{\gamma + 1}{16} M_1^4 \right)} \right] \quad (7)$$

DISCUSSION OF TABLES AND CHARTS

The changes in Mach number, pressure, density, entropy, and the deviation of the flow across the shock wave are presented in tabular form and on the charts. Table I includes solutions of shock-wave equations for stream Mach numbers from $M_1 = 1.05$ to $M_1 = 4.0$ and for a range of angles of shock from a simple Mach wave to an angle normal to the stream (normal shock). For each angle of shock, the ratio of the pressure behind the shock to the pressure in front of the shock, the ratio of the density in front of the shock to the density behind the shock, the deviation of the flow across the shock, and the change in entropy across the shock are given. Solutions of the shock-wave equations which give Mach numbers behind the shock of less than one are included because they are useful in the study of detached shock, internal flows, and similar fields. Table II gives values of the angle of the shock and the Mach number behind the shock for the maximum possible deviation of the flow as well as values of the angle of the shock and the deviation of the flow which gives sonic velocity behind the shock ($M_2 = 1.0$).

The data given in table I were plotted and from these curves values of the angle of the shock and the Mach number behind the shock were taken for even values of the deviation of the flow at Mach numbers in front of the shock from 1.05 to 4.0. These values (read from faired curves) are given in table III and are presented solely to aid in the preparation of large-scale figures similar to figure 2, which shows the variation of the angle of the shock with the Mach number in front of the shock for values of the deviation of the flow from zero to the maximum deviation. In figure 3 the variation of the Mach number behind the shock with the Mach number in front of the shock is shown for the same values of the deviation of the flow. Figure 4 gives the variation of the angle of shock with the pressure ratio p_2/p_1 for a range of

Mach number in front of the shock from $M_1 = 1.05$ to $M_1 = 4.0$. The variation of the maximum deviation of the flow across the shock with the Mach number in front of the shock is shown in figure 5.

Langley Aeronautical Laboratory

National Advisory Committee for Aeronautics
Langley Field, Va., March 31, 1948

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TABLE I

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
 AND CHANGE IN ENTROPY ACROSS SHOCK WAVES

M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{P_1}{P_2}$	$\frac{\Delta S}{\frac{ft^2}{sec^2} \frac{°F}{°F}}$	M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{P_1}{P_2}$	$\frac{\Delta S}{\frac{ft^2}{sec^2} \frac{°F}{°F}}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
1.05	72	15	0	0	1.050	1.000	1.000	0	1.40	86	0	2	22	0.768	1.949	0.6262	51.435
	73	0	0	16	1.037	1.010	.9932	-----		90	0	0	0	.762	1.960	.6239	52.835
	76	0	0	25	1.014	1.044	.9695	.0010		45	35	0	0	1.400	1.000	1.000	0
	79	0	0	33	.991	1.073	.9511	.0454		47	0	1	7	1.361	1.056	.9616	.0168
	82	0	0	32	.973	1.095	.9375	.1284		50	0	3	17	1.279	1.175	.8912	.7618
	85	0	0	23	.961	1.110	.9283	.1739		53	0	5	8	1.211	1.292	.8333	2.773
	88	0	0	10	.954	1.118	.9234	.2035		56	0	6	40	1.114	1.405	.7853	6.944
	90	0	0	-----	.953	1.120	.9225	.2865		59	0	7	52	1.082	1.513	.7453	12.268
1.10	65	23	0	0	1.100	1.000	1.000	0	1.45	62	0	8	45	1.024	1.616	.7120	19.157
	68	0	0	40	1.063	1.047	.9678	.0494		65	0	9	16	.971	1.712	.6843	26.968
	71	0	1	11	1.025	1.095	.9370	.2095		68	0	9	25	.922	1.799	.6613	35.244
	74	0	1	27	.993	1.138	.9120	.3359		71	0	12	35	.878	1.878	.6422	43.735
	77	0	1	31	.965	1.174	.8921	.8437		74	0	8	35	.840	1.946	.6268	51.279
	80	0	1	22	.940	1.202	.8768	.9465		77	0	7	35	.807	2.004	.6145	.58225
	83	0	1	4	.928	1.224	.8658	1.293		80	0	6	2	.785	2.029	.6096	61.135
	86	0	0	39	.918	1.238	.8587	1.601		83	0	4	36	.760	2.086	.5982	68.769
	90	0	0	-----	.912	1.245	.8554	1.828		86	0	2	43	.746	2.109	.5939	71.876
1.15	60	24	0	0	1.150	1.000	1.000	0	1.45	90	0	0	0	.740	2.120	.5918	73.225
	63	0	0	57	1.105	1.058	.9604	.0188		43	36	0	0	1.450	1.000	1.000	0
	66	0	1	47	1.058	1.121	.9217	.2421		44	0	0	21	1.438	1.017	.9880	.0524
	69	0	2	20	1.016	1.178	.8896	.8121		47	0	2	46	1.354	1.145	.9077	.3804
	72	0	2	37	.980	1.229	.8633	1.412		50	0	4	54	1.275	1.273	.8421	2.590
	75	0	2	39	.948	1.273	.8420	2.427		53	0	6	43	1.204	1.398	.7881	6.588
	78	0	2	27	.922	1.310	.8053	3.541		56	0	8	13	1.135	1.519	.7433	12.621
	81	0	2	2	.902	1.339	.8126	4.332		59	0	9	24	1.072	1.636	.7061	20.816
	84	0	1	27	.887	1.359	.8038	5.105		62	0	10	13	1.021	1.746	.6751	29.965
	87	0	0	45	.879	1.372	.7985	5.475		65	0	10	41	.959	1.848	.6492	40.298
	90	0	0	0	.875	1.376	.7968	5.487		68	0	10	46	.909	1.942	.6277	50.600
	56	26	0	0	1.200	1.000	1.000	0		71	0	10	27	.861	2.026	.6100	61.012
	59	0	1	12	1.149	1.068	.9543	.0178		74	0	9	42	.866	2.100	.5956	70.594
	62	0	2	20	1.095	1.143	.9090	.5286		77	0	8	34	.790	2.162	.5841	79.032
	65	0	3	9	1.045	1.213	.8712	1.317		80	0	6	48	.767	2.188	.5795	82.727
	68	0	3	41	1.001	1.278	.8988	1.952		83	0	5	11	.740	2.250	.5690	91.785
	71	0	3	56	.962	1.335	.8140	4.327		86	0	3	3	.727	2.274	.5650	95.562
	74	0	3	53	.988	1.386	.7930	6.135		90	0	0	0	.720	2.286	.5630	97.685
1.20	53	8	0	0	1.250	1.000	1.000	0	1.50	41	49	0	0	1.500	1.000	1.000	0
	54	0	0	31	1.230	1.026	.9815	-----		45	0	2	47	1.405	1.146	.9074	.3339
	57	0	2	4	1.166	1.116	.9249	.1531		48	0	5	5	1.322	1.283	.8373	2.613
	60	0	3	20	1.108	1.201	.8778	.9405		51	0	7	46	1.246	1.419	.7799	7.361
	63	0	4	17	1.055	1.281	.8385	2.517		54	0	8	46	1.174	1.551	.7325	14.752
	66	0	4	55	1.006	1.355	.8057	4.948		57	0	10	10	1.107	1.680	.6932	24.227
	69	0	5	15	.963	1.422	.7786	7.532		60	0	11	49	1.045	1.802	.6605	35.484
	72	0	5	15	.924	1.482	.7633	10.655		63	0	11	49	.986	1.917	.6332	47.745
	75	0	4	57	.891	1.531	.7383	13.550		66	0	12	6	.932	2.024	.6105	60.597
	78	0	4	22	.863	1.578	.7241	16.587		69	0	11	59	.882	2.121	.5916	73.539
	81	0	3	31	.841	1.612	.7134	18.752		72	0	11	25	.837	2.208	.5761	85.625
	84	0	2	28	.826	1.636	.7059	20.695		75	0	10	26	.797	2.286	.5636	96.758
	87	0	1	16	.817	1.651	.7015	22.118		78	0	9	18	.764	2.345	.5538	106.52
	90	0	0	-----	.813	1.656	.7000	22.229		81	0	7	10	.737	2.394	.5463	114.27
1.30	50	17	0	0	1.300	1.000	1.000	0	1.55	40	11	0	0	1.550	1.000	1.000	0
	53	0	1	43	1.235	1.091	.9398	.0207		43	0	2	36	1.461	1.137	.9124	.2618
	56	0	3	20	1.169	1.189	.8841	.9998		46	0	5	5	1.375	1.284	.8370	2.758
	59	0	4	39	1.109	1.282	.8378	2.668		49	0	7	15	1.291	1.430	.7756	.7822
	62	0	5	38	1.053	1.370	.7992	5.357		52	0	9	8	1.219	1.574	.7253	16.209
	65	0	6	18	1.052	1.453	.7670	9.144		55	0	11	55	1.081	1.849	.6190	53.883
	68	0	6	38	.956	1.528	.7403	13.304		61	0	12	47	1.018	1.977	.6201	1.142
	71	0	6	37	.914	1.596	.7182	17.718		64	0	13	18	.960	2.098	.5960	70.187
	74	0	6	16	.878	1.655	.7003	22.297		67	0	13	23	.905	2.208	.5760	86.652
	77	0	5	36	.848	1.705	.6860	26.324		70	0	13	3	.855	2.308	.5595	100.70
	80	0	4	28	.828	1.726	.6803	28.213		73	0	12	15	.811	2.397	.5460	114.74
	83	0	3	26	.804	1.776	.6672	34.762		76	0	10	59	.772	2.472	.5351	127.18
	86	0	2	2	.792	1.795	.6622	34.762		79	0	9	16	.739	2.534	.5266	137.84
	90	0	0	0	.786	1.805	.6598	35.832		82	0	7	7	.714	2.582	.5204	146.19
1.35	47	47	0	0	1.350	1.000	1.000	0	1.60	38	41	0	0	1.600	1.000	1.000	0
	50	0	1	35	1.293	1.081	.9459	.2332		41	0	2	16	1.524	1.119	.9830	.1492
	53	0	3	28	1.221	1.190	.8835	.8012		44	0	4	55	1.433	1.275	.8413	.2321
	56	0	5	2	1.155	1.295	.8319	2.943		47	0	7	16	1.347	1.431	.7753	.8033
	59	0	6	17	1.094	1.396	.7890	6.531		50	0	9	19	1.268	1.586	.7214	17.068
	62	0	7	13	1.037	1.491	.7532	11.222		53	0	11	4	1.193	1.738	.6770	29.336
	65	0	7	49	.958	1.580	.7233	16.531		56	0	12	29	.732	1.886	.6403	44.264
	68	0	8	3	.938	1.661	.6986	22.793		59	0	13	35	1.056	2.027	.6097	51.249
	71	0	7	55	.895	1.734	.6781	28.932		62	0	14	18	.935	2.162	.5812	79.141
	74	0	7	26	.858	1.798	.6615	34.981		65	0	14	38	.934	2.287	.5630	97.414
	77	0	6	36	.826	1.852	.6										

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,

AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec^2) \text{ of } F}$	M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec^2) \text{ of } F}$	
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)					
1.60	74	0	12	55	0.786	2.593	0.5190	148.21	1.85	32	43	0	0	1.850	1.000	1.000	0	
	77	0	11	22	.748	2.669	.5095	162.13		34	0	1	29	1.799	1.082	.9154	.0761	
	80	0	9	21	.722	2.701	.5058	173.06		37	0	4	42	1.687	1.280	.8389	3.029	
	83	0	6	52	.692	2.776	.4971	181.56		40	0	7	37	1.585	1.483	.7560	10.761	
	86	0	4	4	.676	2.805	.4938	187.21		43	0	10	14	1.490	1.691	.6902	25.168	
	90	0	0	0	.668	2.820	.4922	190.04		46	0	12	35	1.400	1.899	.6372	45.795	
										49	0	14	39	1.314	2.108	.5941	71.556	
										52	0	16	26	2.132	2.313	.5588	101.48	
										55	0	17	54	1.153	2.513	.5295	134.09	
										58	0	19	3	1.078	2.705	.5052	171.91	
1.65	37	18	0	0	1.650	1.000	1.000	0	1.90	61	0	19	49	1.007	2.888	.4850	203.15	
	38	0	0	44	1.626	1.037	.9742	1.074		64	0	20	11	.939	3.059	.4681	237.22	
	41	0	3	40	1.526	1.200	.8778	1.074		67	0	20	4	.876	3.217	.4540	269.74	
	45	0	7	5	1.406	1.422	.7789	7.728		70	0	19	27	.817	3.359	.4424	300.20	
	48	0	9	19	1.322	1.587	.7208	16.545		73	0	18	14	.763	3.485	.4329	327.61	
	51	0	11	15	1.243	1.752	.6735	30.582		76	0	16	23	.716	3.593	.4253	351.39	
	54	0	12	53	1.169	1.912	.6343	47.195		79	0	13	52	.676	3.681	.4193	371.09	
	57	0	14	11	1.098	2.067	.6019	66.274		82	0	10	43	.643	3.749	.4150	386.58	
	60	0	15	8	1.032	2.216	.5748	86.781		85	0	7	0	.620	3.796	.4120	397.11	
	63	0	15	42	.969	2.355	.5522	108.07		88	0	2	52	.608	3.821	.4105	402.93	
1.70	66	0	15	51	.910	2.184	.5334	129.27	1.90	90	0	0	0	.606	3.826	.4102	405.62	
	69	0	15	33	.856	2.602	.5179	149.64		93	0	10	26	1.530	1.719	.6822	27.594	
	72	0	14	45	.806	2.706	.5051	168.66		96	0	12	51	1.437	1.939	.6284	50.345	
	75	0	13	27	.762	2.797	.4947	185.57		98	0	15	28	1.083	.9449	.1927		
	78	0	11	36	.725	2.872	.4866	200.02		101	0	0	0	1.948	2.848	.2.962		
	81	0	9	15	.695	2.932	.4804	211.67		104	0	4	46	1.734	2.289	.8348		
	84	0	6	27	.672	2.975	.4761	220.27		107	0	7	45	1.628	2.590	.5194		
	87	0	3	19	.659	3.001	.4736	225.50		110	0	10	26	1.501	2.796	.4949		
	90	0	0	0	.654	3.010	.4728	227.26		113	0	12	55	1.530	2.992	.4745		
										116	0	15	35	1.032	2.992	.4745		
1.70	36	2	0	0	1.700	1.000	1.000	0	1.95	119	0	0	0	1.900	1.000	1.000	0	
	37	0	1	2	1.665	1.055	.9628	.1818		122	0	1	28	1.848	1.083	.9449	.1927	
	40	0	4	3	1.562	1.226	.8646	1.610		125	0	16	52	1.265	2.377	.5489	111.52	
	44	0	7	34	1.439	1.460	.7642	9.630		128	0	18	27	1.184	2.590	.5194	147.61	
	47	0	9	53	1.350	1.637	.7058	20.803		131	0	19	41	1.106	2.796	.4949	185.39	
	50	0	11	54	1.272	1.812	.6580	36.407		134	0	20	35	1.032	2.992	.4745	223.75	
	53	0	13	37	1.195	1.984	.6188	55.689		137	0	21	56	.962	3.177	.4574	261.49	
	56	0	15	0	1.122	2.151	.5862	77.555		140	0	21	8	.895	3.348	.4433	297.92	
	59	0	16	4	1.052	2.311	.5591	101.14		143	0	20	39	.833	3.504	.4335	331.73	
	62	0	16	44	.987	2.462	.5365	125.46		146	0	22	6	.776	3.613	.4219	362.52	
1.75	65	0	17	1	.925	2.602	.5177	149.93	1.95	149	0	17	55	.725	3.763	.4141	389.54	
	68	0	16	49	.867	2.732	.5021	173.35		152	0	15	31	.681	3.863	.4079	412.39	
	71	0	16	8	.815	2.848	.4892	195.27		155	0	12	26	.645	3.942	.4033	430.64	
	74	0	14	55	.768	2.949	.4787	215.00		158	0	8	43	.618	3.999	.4001	443.81	
	77	0	13	9	.726	3.034	.4704	232.17		161	0	4	30	.601	4.033	.3981	452.76	
	80	0	10	49	.698	3.070	.4670	239.48		164	0	0	0	.596	4.045	.3975	454.52	
	83	0	7	58	.666	3.155	.4594	256.95		167	0	10	21	.517	4.177	.3965		
	86	0	4	43	.649	3.189	.4564	263.86		170	0	12	54	.581	4.207	.3958		
	90	0	0	0	.641	3.205	.4550	267.45		173	0	1	23	1.901	2.079	.9471	.1650	
										176	0	4	45	1.782	2.193	.8328	.3.053	
1.75	34	51	0	0	1.750	1.000	1.000	0	2.00	179	0	0	7	48	1.674	1.515	.7449	12.620
	36	0	1	16	1.707	1.068	.9543	.1680		182	0	10	34	1.573	1.743	.6758	29.754	
	39	0	4	21	1.602	1.248	.8537	2.002		185	0	13	3	1.478	1.974	.6208	54.438	
	42	0	7	7	1.505	1.433	.7744	5.679		188	0	15	17	1.386	2.206	.5764	85.426	
	45	0	9	35	1.411	1.620	.7109	19.498		191	0	17	14	1.299	2.437	.5401	121.25	
	48	0	11	47	1.330	1.807	.6594	35.868		194	0	18	54	1.216	2.663	.5103	160.67	
	51	0	13	41	1.249	1.991	.6172	56.587		197	0	20	15	1.136	2.882	.4855	200.76	
	54	0	15	17	1.172	2.172	.5624	80.500		200	0	21	16	1.059	3.098	.4649	251.41	
	57	0	16	33	1.099	2.346	.5535	106.72		203	0	21	54	.986	3.292	.4478	285.66	
	60	0	17	29	1.029	2.513	.5295	131.17		206	0	22	6	.917	3.477	.4335	325.83	
1.80	63	0	18	6	.963	2.670	.5094	161.99	2.00	209	0	20	54	.791	3.799	.4118	397.89	
	66	0	17	43	.844	2.947	.4789	218.78		212	0	19	23	.737	3.935	.4038	430.70	
	69	0	16	48	.792	3.065	.4675	238.48		215	0	17	9	.689	4.045	.3975	454.48	
	72	0	15	19	.745	3.167	.4583	259.45		218	0	14	11	.649	4.136	.3962	475.59	
	75	0	13	14	.705	3.252	.4511	277.31		221	0	10	31	.617	4.204	.3891	491.55	
	78	0	10	34	.672	3.318	.4566	291.52		224	0	6	15	.596	4.248	.3869	501.92	
	81	0	7	23	.648	3.367	.4418	301.90		227	0	0	0	.586	4.270	.3858	507.06	
	84	0	3	48	.633	3.396	.4395	308.21		230	0	0	0	0	0	0		
	87	0	0	0	.628	3.406	.4388	310.30		233	0	0	0	0	0	0		
	90	0	0	0	.617	3.613	.4239	316.00		236	0	0	0	0	0	0		

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,

AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{Ft^2/sec^2}$ $\frac{^{\circ}F}{^{\circ}F}$	M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{Ft^2/sec^2}$ $\frac{^{\circ}F}{^{\circ}F}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
2.05	29	12	0	0	2.050	1.000	1.000	0	2.20	72	0	24	18	0.761	1.4941	0.3570	667.37
	30	0	1	0	2.013	1.059	.9598	.0336		75	0	22	20	.701	5.122	.3512	706.33
	33	0	4	32	1.886	1.288	.8351	2.667		78	0	19	31	.650	5.234	.3466	738.68
	36	0	7	43	1.771	1.527	.7406	13.094		81	0	15	47	.607	5.342	.3431	764.31
	39	0	10	37	1.664	1.775	.6673	32.672		84	0	11	9	.574	5.418	.3407	782.94
	42	0	13	14	1.563	2.029	.6095	61.108		87	0	5	47	.554	5.465	.3393	794.15
	45	0	15	37	1.467	2.284	.5632	97.103		90	0	0	0	.547	5.480	.3388	798.01
	48	0	17	43	1.374	2.541	.5257	138.82									
	51	0	19	34	1.286	2.795	.4950	185.00	2.25	26	23	0	0	2.250	1.000	1.000	0
	54	0	21	7	1.200	3.042	.4696	233.68		29	0	3	20	2.122	1.222	.8670	1.226
	57	0	22	22	1.118	3.282	.4486	283.54		32	0	6	48	1.990	1.492	.7528	11.034
	60	0	23	15	1.040	3.511	.4310	333.09		35	0	9	58	1.869	1.776	.6670	32.873
	63	0	23	44	.965	3.726	.4164	381.21		38	0	12	51	1.755	2.072	.6009	66.707
	66	0	23	46	.894	3.925	.4043	426.60		41	0	15	29	1.648	2.376	.5491	111.22
	69	0	23	14	.828	4.107	.3942	468.77		44	0	17	53	1.544	2.683	.5078	164.30
	72	0	22	6	.766	4.268	.3859	506.65		47	0	20	2	1.445	2.992	.4744	223.75
	75	0	20	15	.711	4.408	.3792	539.66		50	0	21	57	1.349	3.301	.4472	289.97
	78	0	17	37	.663	4.524	.3739	567.34		53	0	23	36	1.257	3.601	.4247	352.90
	81	0	14	11	.623	4.616	.3699	589.36		56	0	24	57	1.168	3.893	.4062	419.20
	84	0	9	58	.594	4.683	.3671	605.33		59	0	25	58	1.083	4.173	.3907	481.09
	87	0	5	10	.575	4.723	.3655	614.90		62	0	26	36	1.001	4.428	.3778	546.70
	90	0	0	0	.569	4.736	.3650	618.04		65	0	26	47	.923	4.685	.3671	605.74
										68	0	26	27	.849	4.911	.3581	660.25
2.10	28	26	0	0	2.100	1.000	1.000	0	71	0	25	28	.781	5.114	.3508	709.13	
	29	0	0	43	2.073	1.043	.9706	.0119		74	0	23	45	.718	5.291	.3448	751.96
	32	0	4	19	1.941	1.278	.8395	2.431		77	0	21	10	.663	5.441	.3400	788.50
	35	0	7	35	1.823	1.526	.7410	13.010		80	0	17	40	.615	5.562	.3364	817.71
	38	0	10	32	1.712	1.784	.6652	33.503		83	0	13	12	.578	5.632	.3337	839.54
	41	0	13	14	1.608	2.048	.6057	63.611		86	0	7	54	.553	5.711	.3321	853.92
	44	0	15	40	1.509	2.316	.5580	101.86		90	0	0	0	.541	5.740	.3313	860.86
	47	0	17	51	1.414	2.585	.5199	146.68									
	50	0	19	47	1.323	2.852	.4887	196.12	2.30	25	46	0	0	2.300	1.000	1.000	0
	53	0	21	26	1.235	3.115	.4629	248.57		28	0	2	53	2.186	1.194	.8814	.7943
	56	0	22	47	1.151	3.369	.4416	302.28		31	0	6	27	2.050	1.472	.7601	9.925
	59	0	23	48	1.070	3.614	.4238	355.83		34	0	9	40	1.925	1.763	.6704	31.579
	62	0	24	26	.992	3.844	.4090	407.99		37	0	12	37	1.808	2.069	.6016	66.296
	65	0	24	37	.918	4.060	.3967	457.66		40	0	15	18	1.697	2.383	.5479	112.53
	68	0	24	16	.849	4.256	.3865	503.91		43	0	17	46	1.591	2.704	.5053	168.02
	71	0	23	20	.784	4.433	.3780	545.66		46	0	20	0	1.489	3.027	.4711	229.53
	74	0	21	41	.726	4.588	.3712	582.41		49	0	21	59	1.390	3.349	.4432	294.87
	77	0	19	16	.674	4.718	.3657	613.80		52	0	23	43	1.295	3.666	.4203	367.61
	80	0	16	0	.630	4.823	.3615	639.05		55	0	25	10	1.212	3.975	.4014	438.08
	83	0	11	54	.596	4.902	.3585	657.99		58	0	26	19	1.116	4.272	.3857	507.53
	86	0	7	6	.573	4.953	.3565	670.39		61	0	27	6	1.032	4.555	.3726	574.68
	90	0	0	0	.561	4.978	.3556	676.56		64	0	27	26	.951	4.819	.3617	638.07
										67	0	27	17	.874	5.063	.3526	696.90
2.15	27	43	0	0	2.150	1.000	1.000	0	70	0	26	32	.803	5.283	.3451	750.17	
	28	0	0	22	2.139	1.022	.9846	.0020		73	0	25	3	.737	5.478	.3389	797.33
	31	0	4	4	1.999	1.265	.8458	2.102		76	0	22	45	.678	5.614	.3340	837.74
	34	0	7	23	1.876	1.520	.7432	12.711		79	0	19	29	.626	5.781	.3301	870.83
	37	0	10	25	1.763	1.787	.6644	33.790		82	0	15	14	.585	5.886	.3273	896.25
	40	0	13	10	1.656	2.062	.6030	65.350		85	0	10	3	.555	5.958	.3254	913.86
	43	0	15	40	1.554	2.342	.5542	105.89		88	0	4	9	.537	5.998	.3244	923.31
	46	0	17	56	1.456	2.624	.5151	153.61		90	0	0	0	.534	6.005	.3242	925.18
	49	0	21	41	1.272	3.182	.4570	262.52	2.35	25	11	0	0	2.350	1.000	1.000	0
	52	0	23	8	1.185	3.452	.4353	320.29		27	0	2	24	2.254	1.161	.8988	.4871
	55	0	24	16	1.101	3.712	.4173	378.01		30	0	6	1	2.112	1.444	.7702	8.584
	58	0	25	2	1.028	3.959	.4023	434.36		33	0	9	19	1.983	1.745	.6754	29.851
	61	0	25	22	.944	4.190	.3898	488.25		36	0	12	20	1.862	2.059	.6034	65.009
	64	0	25	12	.872	4.402	.3794	538.44		39	0	5	15	1.748	2.385	.5477	112.76
	67	0	24	28	.804	4.596	.3708	584.34		42	0	17	36	1.639	2.718	.5037	170.62
	70	0	23	3	.742	4.765	.3638	625.23		45	0	19	54	1.534	3.055	.4684	236.22
	73	0	20	51	.686	4.911	.3581	660.21		48	0	21	58	1.433	3.392	.4399	306.99
	76	0	17	47	.639	5.030	.3537	688.94		51	0	23	47	1.335	3.725	.4165	380.88
	79	0	13	51	.600	5.122	.3505	711.13		54	0	25	20	1.241	4.050	.3972	455.56
	82	0	9	6	.572	5.185	.3483	726.45		57	0	26	35	1.151	4.365	.3812	529.42
	85	0	3	45	.557	5.220	.3472	734.85		60	0	27	30	1.064	4.666	.3678	601.08
	88	0	0	0	.554	5.226	.3469	736.44		63	0	28	---	.980	4.949	.3567	669.34
										66	0	28	2	.901	5.211	.3475	752.57
2.20	27	2	0	0	2.200	1.000	1.000	0	69	0	27	29	.827	5.449	.3398	.790.44	
	30	0	3	43	2.059	1.245	.8553	1.720		72	0	26	15	.757	5.651	.3335	841.84
	33	0	7	7	1.932	1.508	.7471	11.985		75	0	24	13	.694	5.845	.3284	886.40
	36	0	10	13	1.815	1.784	.6650	33.590		78	0	21	14	.639	5.998	.3244	923.36
	39	0	13	2	1.705	2.070	.6014	66.336		81	0	17	14	.593	6.119	.3213	952.54
	42	0	15	36	1.600	2.362	.5512	109.03		84	0	12	13	.558	6.206	.3192	973.52
	45	0	17	56	1.499	2.657	.5110	159.44		87	0						

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
 AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec^2) \text{ OF}}$	M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec^2) \text{ OF}}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
2.40	41	0	17	23	1.688	2.726	0.5028	172.07	2.55	80	0	20	33	0.594	7.191	.2988	1208.54
	44	0	19	45	1.581	3.076	.4665	240.56		83	0	15	29	.552	7.307	.2967	1235.73
	47	0	21	53	1.492	3.428	.4322	329.56		86	0	9	19	.523	7.383	.2954	1253.59
	50	0	23	47	1.376	3.777	.4132	392.62		90	0	0	0	.508	7.420	.2948	1262.20
	53	0	25	26	1.280	4.120	.3935	471.78		22	37	0	0	2.600	1.000	1.000	0
	56	0	26	47	1.187	4.452	.3771	550.07		25	0	3	13	2.460	1.242	.8568	1.650
	59	0	27	49	1.097	4.771	.3636	626.45		28	0	6	53	2.303	1.572	.7260	15.975
	62	0	28	29	1.011	5.072	.3522	699.18		31	0	10	14	2.164	1.927	.6310	48.795
	65	0	28	41	.929	5.353	.3428	767.25		34	0	13	17	2.029	2.299	.5609	99.355
	68	0	28	20	.852	5.610	.3349	829.37		37	0	16	5	1.903	2.690	.5070	165.41
	71	0	27	21	.780	5.841	.3285	885.40		40	0	18	41	1.783	3.092	.4650	213.95
	74	0	25	35	.713	6.043	.3232	934.25		43	0	21	4	1.667	3.502	.4317	331.00
	77	0	22	53	.654	6.213	.3190	975.29		46	0	23	15	1.556	3.914	.4049	424.05
	80	0	19	10	.604	6.350	.3158	1008.42		49	0	25	13	1.448	4.326	.3831	520.20
	83	0	14	23	.564	6.454	.3135	1033.08		52	0	26	57	1.344	4.731	.3656	616.90
	86	0	8	38	.537	6.521	.3120	1049.17		55	0	28	25	1.244	5.126	.3504	712.07
	90	0	0	0	.523	6.554	.3113	1057.03		58	0	29	35	1.148	5.505	.3381	804.08
2.45	24	5	0	0	2.450	1.000	1.000	0	2.60	61	0	30	24	1.056	5.866	.3278	891.44
	25	0	1	15	2.398	1.084	.9439	.0919		64	0	30	47	.968	6.205	.3193	1000.97
	28	0	5	23	2.243	1.377	.7965	.5580		67	0	30	40	.891	6.516	.3121	1047.96
	31	0	8	30	2.105	1.692	.6896	.25181		70	0	29	55	.803	6.798	.3063	1117.32
	34	0	11	37	1.976	2.023	.6106	.6395		73	0	28	24	.733	7.046	.3015	1174.12
	37	0	14	29	1.855	2.370	.5500	.110.28		76	0	25	57	.666	7.259	.2976	1224.31
	40	0	17	8	1.740	2.727	.5027	172.27		79	0	22	25	.609	7.433	.2946	1265.36
	43	0	19	33	1.629	3.091	.4651	243.52		82	0	17	40	.562	7.567	.2924	1296.76
	46	0	21	45	1.522	3.457	.4350	321.33		85	0	11	44	.527	7.660	.2909	1318.12
	49	0	23	44	1.419	3.822	.4104	403.01		88	0	4	52	.508	7.711	.2901	1329.92
	52	0	25	28	1.320	4.182	.3902	486.38		90	0	0	0	.504	7.720	.2899	1332.26
	55	0	26	55	1.224	4.532	.3736	569.31		22	10	0	0	2.650	1.000	1.000	0
	58	0	28	4	1.132	4.870	.3597	650.28		24	0	2	31	2.538	1.189	.8839	-----
	61	0	28	52	1.043	5.190	.3481	727.72		27	0	6	17	2.375	1.522	.7424	12.917
	64	0	29	14	.959	5.491	.3385	800.43		30	0	9	41	2.228	1.882	.6413	43.790
	67	0	29	6	.879	5.767	.3305	867.49		33	0	12	48	2.091	2.264	.5667	93.992
	70	0	28	20	.804	6.017	.3239	928.03		36	0	15	40	1.96	2.664	.5101	160.88
	73	0	26	50	.734	6.238	.3185	981.10		39	0	18	19	1.838	3.078	.4663	241.13
	76	0	24	26	.671	6.427	.3141	1026.47		42	0	20	45	1.718	3.502	.4317	331.24
	79	0	21	2	.616	6.582	.3107	1063.68		45	0	23	0	1.604	3.930	.4040	427.82
	82	0	16	31	.572	6.701	.3082	1092.28		48	0	25	2	1.493	4.358	.3815	527.95
	85	0	10	56	.540	6.783	.3066	1111.86		51	0	26	51	1.387	4.782	.3632	629.24
	88	0	4	31	.521	6.828	.3057	1122.61		54	0	28	24	1.284	5.196	.3480	729.13
	90	0	0	0	.518	6.836	.3055	1124.64		57	0	29	41	1.186	5.596	.3354	826.05
2.50	23	35	0	0	2.500	1.000	1.000	0	2.70	60	0	30	38	1.091	5.978	.3249	918.56
	24	0	39	27	2.477	1.040	.9726	.0099		63	0	31	11	1.000	6.338	.3161	1005.37
	27	0	4	29	2.318	1.336	.8136	.4218		66	0	31	15	.914	6.671	.3089	1085.28
	30	0	8	0	2.169	1.656	.7000	.22.359		69	0	30	44	.832	6.974	.3028	1157.32
	33	0	11	11	2.036	1.996	.6162	.57.183		72	0	29	29	.756	7.244	.2979	1221.07
	36	0	14	7	1.911	2.353	.5526	.107.72		75	0	27	22	.687	7.477	.2939	1275.77
	39	0	16	49	1.793	2.721	.5033	171.33		78	0	24	11	.625	7.672	.2907	1321.10
	42	0	19	18	1.679	3.098	.4645	245.26		81	0	19	47	.573	7.826	.2883	1356.78
	45	0	21	34	1.569	3.479	.4333	326.23		84	0	14	8	.534	7.937	.2866	1382.27
	48	0	23	37	1.463	3.860	.4081	411.85		87	0	7	24	.508	8.004	.2857	1397.83
	51	0	25	26	1.361	4.237	.3874	499.41		90	0	0	0	.500	8.026	.2853	1402.90
	54	0	26	59	1.263	4.606	.3704	586.97		59	0	30	47	1.127	6.082	.3323	826.05
	57	0	28	15	1.168	4.962	.3562	672.66		62	0	31	30	1.034	6.464	.3133	1035.46
	60	0	29	11	1.077	5.302	.3444	754.91		65	0	5	37	2.448	1.468	.7615	9.908
	63	0	29	42	.990	5.622	.3346	832.63		68	0	9	6	2.296	1.832	.6530	38.562
	66	0	29	45	.907	5.919	.3264	904.33		72	0	15	12	2.155	2.222	.5737	87.712
	69	0	29	13	.829	6.189	.3197	969.50		75	0	17	54	1.893	3.057	.4683	236.84
	72	0	27	58	.756	6.429	.3141	1027.15		78	0	17	48	1.771	3.494	.4323	329.61
	75	0	25	53	.690	6.637	.3096	1076.97		81	0	22	42	1.653	3.937	.4036	429.60
	78	0	22	47	.621	6.810	.3060	1118.26		84	0	24	48	1.540	4.382	.3804	533.67
	81	0	18	34	.582	6.947	.3033	1150.70		87	0	26	42	1.431	4.824	.3615	639.51
	84	0	13	13	.545	7.045	.3015	1174.11		90	0	28	21	1.326	5.258	.3459	744.16
	87	0	6	58	.519	7.105	.3004	1188.32		53	0	29	44	1.225	5.679	.3330	846.19
	90	0	0	0	.513	7.125	.3000	1193.01		56	0	30	47	1.127	6.082	.3223	943.73
2.55	23	5	0	0	2.550	1.000	1.000	0	2.75	59	0	0	0	2.750	1.000	1.000	0
	26	0	3	53	2.385	1.291	.8336	2.876		62	0	31	45	.945	6.820	.3058	1120.43
	29	0	7	28	2.235	1.616	.7119	19.290		65	0	31	26	.861	7.145	.2996	1197.59
	32	0	10	44	2.098	1.964	.6230	.53.132		68	0	31	28	.782	7.437	.2945	1266.21
	35	0	13	43	1.969	2.329	.5562	.104.01		71	0	30	48	.647	7.908	.2871	1375.64
	38	0	16	29	1.847	2.709	.5084	.169.05		74	0	28	38	.709	7.692	.2904	1325.55
	41	0	19	1</td													

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
 AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(\text{ft}^2/\text{sec}^2)} \frac{c_F}{c_F}$	M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(\text{ft}^2/\text{sec}^2)} \frac{c_F}{c_F}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
2.75	40	0	20	1	1.826	3.479	0.4333	326.03	2.90	66	0	33	20	0.926	8.022	0.2854	1401.94
	43	0	22	23	1.705	3.937	.4036	429.46		69	0	32	52	.839	8.385	.2804	1484.81
	46	0	24	32	1.588	4.399	.3796	537.44		72	0	31	38	.758	8.708	.2762	1557.77
	49	0	26	30	1.477	4.859	.3601	647.59		75	0	29	29	.684	8.988	.2729	1620.09
	52	0	28	14	1.368	5.312	.3441	757.25		78	0	26	11	.618	9.221	.2702	1671.56
	55	0	29	42	1.264	5.754	.3309	864.19		81	0	21	34	.562	9.405	.2682	1711.92
	58	0	30	54	1.165	6.179	.3199	966.93		84	0	15	30	.519	9.538	.2668	1740.89
	61	0	31	44	1.069	6.583	.3107	1063.86		87	0	8	9	.491	9.618	.2660	1758.18
	64	0	32	9	.977	6.961	.3031	1158.98		90	0	0	0	.481	9.645	.2658	1764.28
	67	0	32	3	.890	7.309	.2967	1236.19	2.95	19	49	0	0	2.95	1.000	1.000	0
	70	0	31	19	.808	7.624	.2914	1309.77		20	0	0	16	2.936	1.021	.9853	.000
	73	0	29	47	.732	7.902	.2872	1374.26		23	0	4	23	2.735	1.383	.7939	5.986
	76	0	27	19	.663	8.140	.2837	1428.95		26	0	8	2	2.559	1.784	.6650	33.698
	79	0	23	40	.603	8.335	.2810	1473.39		29	0	11	21	2.399	2.220	.5741	87.349
	82	0	18	44	.553	8.486	.2790	1507.35		32	0	14	24	2.248	2.684	.5077	164.62
	85	0	12	29	.517	8.589	.2777	1531.02		35	0	16	53	2.090	3.174	.4577	260.81
	88	0	5	11	.496	8.646	.2770	1543.49		38	0	19	52	1.969	3.682	.4193	371.32
	90	0	0	0	.492	8.656	.2769	1546.11		41	0	22	19	1.838	4.203	.3892	491.51
	2.80	20	56	0	0	2.800	1.000	0	3.00	44	0	24	35	1.712	4.732	.3651	617.38
	21	0	0	7	2.795	1.008	.9943	.0000		47	0	26	39	1.590	5.264	.3457	745.74
	24	0	4	11	2.604	1.347	.8091	4.353		50	0	28	32	1.474	5.791	.3299	873.54
	27	0	7	50	2.439	1.719	.6824	27.389		53	0	30	12	1.360	6.309	.3168	998.57
	30	0	11	8	2.288	2.120	.5918	73.021		56	0	31	36	1.255	6.81	.3060	1118.74
	33	0	14	10	2.145	2.547	.5250	139.75		59	0	32	42	1.152	7.293	.2970	1232.49
	36	0	16	59	2.011	2.994	.4743	223.81		62	0	33	26	1.053	7.749	.2895	1338.98
	39	0	19	36	1.882	3.456	.4350	320.95		65	0	33	44	.959	8.173	.2832	1436.41
	42	0	22	0	1.757	3.929	.4041	427.22		68	0	33	28	.870	8.561	.2761	1524.82
	45	0	24	14	1.638	4.407	.3792	539.17		71	0	32	31	.786	8.910	.2738	1602.73
	48	0	26	15	1.523	4.884	.3591	653.23		74	0	30	41	.708	9.215	.2703	1670.38
	51	0	28	4	1.412	5.358	.3426	768.04		77	0	27	48	.638	9.473	.2675	1726.62
	54	0	29	38	1.305	5.820	.3291	880.12		80	0	25	54	.568	9.680	.2654	1771.85
	57	0	30	55	1.203	6.267	.3178	988.02		83	0	17	57	.529	9.836	.2639	1805.48
	60	0	31	54	1.105	6.693	.3084	1090.41		86	0	10	54	.495	9.937	.2629	1827.06
	63	0	32	28	1.011	7.095	.3005	1185.63		90	0	0	0	.478	9.986	.2624	1837.61
	66	0	32	34	.921	7.467	.2940	1272.94	3.00	41	0	22	38	1.850	4.352	.3818	526.62
	69	0	32	4	.836	7.805	.2886	1351.78		44	0	24	54	1.723	4.900	.3585	657.70
	72	0	30	50	.757	8.107	.2842	1421.21		47	0	20	46	1.960	1.062	.9582	.0375
	75	0	28	41	.685	8.367	.2806	1480.64		50	0	4	50	2.758	1.436	.7731	8.219
	78	0	25	26	.620	8.585	.2778	1529.76		53	0	0	8	2.581	1.851	.6485	10.430
	81	0	20	53	.566	8.756	.2756	1568.17		56	0	30	31	1.369	6.531	.3118	99.591
	84	0	14	59	.524	8.880	.2741	1595.87		59	0	31	55	1.261	7.050	.3014	1175.15
	87	0	7	51	.498	8.955	.2732	1612.60		62	0	33	2	1.156	7.548	.2927	1292.26
	90	0	0	0	.488	8.980	.2730	1618.43		65	0	33	46	1.057	8.019	.2854	1401.11
	2.85	20	32	0	0	2.850	1.000	0	3.05	68	0	24	54	1.723	4.900	.3585	657.70
	23	0	3	24	2.687	1.280	.8386	2.366		71	0	26	59	1.600	5.450	.3398	790.44
	26	0	7	8	2.514	1.654	.7005	21.972		74	0	28	52	1.483	5.995	.3244	922.61
	29	0	10	31	2.358	2.061	.6631	65.049		77	0	30	31	1.369	6.531	.3118	1051.46
	32	0	13	37	2.211	2.494	.5320	130.80		80	0	31	55	1.261	7.050	.3014	1175.15
	35	0	16	28	2.072	2.951	.4785	215.30		83	0	33	2	1.156	7.548	.2927	1292.26
	38	0	19	8	1.939	3.425	.4373	314.22		86	0	33	46	1.057	8.019	.2854	1401.11
	41	0	21	36	1.811	3.912	.4050	423.51		89	0	34	4	.961	8.458	.2794	1501.26
	44	0	23	53	1.689	4.406	.3793	539.08		92	0	33	49	.871	8.866	.2744	1591.45
	47	0	25	58	1.570	4.902	.3585	657.96		95	0	32	52	.787	9.221	.2702	1671.32
	50	0	27	51	1.457	5.394	.3415	777.09		98	0	31	3	.708	9.536	.2669	1740.34
	53	0	29	31	1.348	5.878	.3275	894.04		101	0	28	9	.637	9.806	.2642	1797.98
	56	0	30	54	1.243	6.346	.3159	1007.32		104	0	23	56	.576	10.017	.2621	1844.15
	59	0	31	59	1.142	6.796	.3063	1059.43		107	0	18	14	.527	10.178	.2606	1878.23
	62	0	32	43	1.045	7.221	.2983	1215.30		110	0	11	4	.493	10.283	.2597	1900.44
	65	0	32	59	.993	7.617	.2916	1308.29		113	0	0	0	.475	10.333	.2593	1911.41
	68	0	32	42	.866	7.980	.2860	1392.15	3.05	116	0	0	0	3.050	1.000	1.0000	0
	71	0	31	44	.784	8.305	.2814	1466.51		119	0	1	15	2.984	1.103	.9324	.0553
	74	0	29	55	.708	8.590	.2777	1531.03		122	0	5	16	2.782	1.490	.7534	11.010
	77	0	27	2	.640	8.830	.2747	1584.93		125	0	8	51	2.603	1.919	.6328	47.836
	80	0	22	55	.581	9.024	.2724	1628.01		128	0	12	7	2.440	2.381	.5478	112.68
	83	0	17	24	.534	9.169	.2708	1660.18		131	0	15	7	2.285	2.881	.4856	201.79
	86	0	10	32	.501	9.264	.2698	1681.05		134	0	17	55	2.139	3.404	.4389	309.65
	90	0	0	0	.485	9.310	.2693	1691.11		137	0	20	31	1.998	3.947	.4030	431.82
	2.90	20	10	0	0	2.900	1.000	0	3.05	141	0	22	57	1.863	4.505	.3748	.562.70
	21	0	1	12	2.842	1.093	.9382	.1255		144	0	25	13	1.734	5.071	.3523	698.86
	24	0	5	11	2.651	1.457	.7656	9.189		147	0	27	17	1.610	5.639	.3341	836.42
	27	0	8	45	2												

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
AND CHANGE IN ENTHALPY ACROSS SHOCK WAVES - Continued

M_1	ϵ		δ		M_2	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec)^2}$	M_1	ϵ		δ		M_2	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec)^2}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
3.10	18	49	0	0	3.100	1.000	1.0000	0	3.25	35	55	19	7	2.202	3.888	0.4065	404.14
	20	0	1	43	3.010	1.145	.9079	.2717		38	0	21	41	2.054	4.504	.3748	562.45
	23	0	5	41	2.806	1.545	.7346	14.353		41	0	24	5	1.912	5.137	.3500	714.81
	26	0	9	14	2.621	1.988	.6179	.55.921		44	0	26	19	1.776	5.780	.3302	870.56
	29	0	12	28	2.460	2.469	.5356	126.62		47	0	28	23	1.646	6.425	.3142	1025.95
	32	0	15	27	2.303	2.982	.4754	221.57		50	0	30	15	1.521	7.065	.3011	1178.65
	35	0	18	14	2.155	3.522	.4302	335.62		53	0	31	56	1.429	7.693	.2904	1325.99
	38	0	20	50	2.012	4.083	.3954	463.34		56	0	33	21	1.288	8.303	.2815	1466.21
	41	0	23	15	1.876	4.659	.3681	.599.73		59	0	34	29	1.179	8.888	.2740	1597.61
	44	0	25	30	1.745	5.244	.3464	.740.57		62	0	35	16	1.074	9.441	.2679	1719.76
	47	0	27	35	1.619	5.830	.3288	.882.83		65	0	35	36	.975	9.956	.2627	1831.08
	50	0	29	27	1.498	6.413	.3144	1023.28		68	0	35	23	.880	10.427	.2585	1931.61
	53	0	31	7	1.383	6.985	.3026	1159.52		71	0	34	29	.792	10.851	.2549	2019.32
	56	0	32	32	1.272	7.539	.2928	1290.06		74	0	32	41	.709	11.220	.2520	2095.10
	59	0	33	39	1.166	8.071	.2847	1413.16		77	0	29	45	.635	11.533	.2498	2158.45
	62	0	34	24	1.064	8.574	.2779	1527.55		80	0	25	25	.570	11.785	.2480	2208.96
	65	0	34	44	.967	9.043	.2722	1636.03		83	0	19	27	.517	11.974	.2467	2246.27
	68	0	34	29	.875	9.172	.2675	1726.51		86	0	11	52	.481	12.097	.2459	2270.58
	71	0	33	33	.789	9.857	.2637	1809.79		89	0	3	4	.463	12.153	.2456	2281.43
	74	0	31	45	.708	10.193	.2605	1881.58		90	0	0	0	.462	12.156	.2456	2282.42
	77	0	28	50	.636	10.478	.2580	1941.57									
3.13	80	0	24	33	.573	10.703	.2561	1989.35	3.30	17	35	0	0	3.300	1.000	1.0000	0
	83	0	18	44	.523	10.879	.2547	2025.11		18	0	0	32	3.268	1.047	.9680	.0000
	86	0	11	24	.488	10.991	.2538	2048.19		20	0	3	22	3.110	1.320	.8208	3.555
	89	0	2	56	.471	11.042	.2534	2058.69		23	0	7	10	2.901	1.773	.6679	32.532
	90	0	0	0	.470	11.045	.2534	2059.38		26	0	10	36	2.716	2.275	.5648	95.417
	3.15	18	31	0	0	3.150	1.000	1.0000	0	29	0	13	14	2.539	2.820	.4922	189.79
	20	0	2	9	3.034	1.188	.8845	.6698		32	0	16	39	2.374	3.401	.4392	309.14
	23	0	6	5	2.830	1.601	.7167	17.961		35	0	19	23	2.218	4.013	.3992	447.00
	26	0	9	36	2.648	2.058	.6037	64.868		38	0	21	56	2.067	4.649	.3685	607.17
	29	0	12	48	2.479	2.554	.5240	141.23		41	0	24	20	1.924	5.302	.3444	754.63
	32	0	15	46	2.321	3.084	.4657	242.24		44	0	26	34	1.787	5.964	.3252	915.30
	35	0	18	32	2.170	3.642	.4219	362.33		47	0	28	38	1.664	6.629	.3097	1074.96
	38	0	21	8	2.026	4.222	.3882	.495.57		50	0	30	30	1.529	7.289	.2971	1231.29
	41	0	23	32	1.888	4.816	.3618	637.43		53	0	32	11	1.409	7.937	.2866	1382.34
	44	0	25	47	1.755	5.120	.3407	783.40		56	0	33	36	1.293	8.566	.2780	1525.56
	47	0	27	51	1.628	6.026	.3237	929.92		59	0	34	44	1.183	9.169	.2708	1660.07
	50	0	29	44	1.506	6.627	.3098	1074.44		62	0	35	31	1.078	9.738	.2648	1784.22
	53	0	31	24	1.390	7.218	.2983	1214.69		65	0	35	52	.977	10.270	.2598	1897.56
	56	0	32	49	1.277	7.790	.2888	1348.37		68	0	35	40	.882	10.756	.2557	1999.51
	59	0	33	56	1.170	8.340	.2810	1474.46		71	0	34	46	.793	11.192	.2523	2089.26
	62	0	34	42	1.067	8.859	.2744	1591.35		74	0	32	58	.710	11.573	.2495	2166.48
	65	0	35	2	.970	9.343	.2689	1698.35		77	0	30	2	.634	11.896	.2473	2230.92
	68	0	34	48	.877	9.786	.2614	1794.61		80	0	25	41	.569	12.156	.2456	2282.41
	71	0	33	53	.790	10.183	.2606	1879.44		83	0	19	41	.516	12.350	.2443	2320.12
	74	0	32	4	.709	10.531	.2575	1952.69		86	0	12	1	.479	12.477	.2436	2344.95
	77	0	29	8	.636	10.825	.2551	2013.81		89	0	3	6	.461	12.535	.2432	2356.21
	80	0	24	51	.572	11.061	.2533	2062.67	3.35	90	0	0	0	.460	12.538	.2432	2356.67
3.20	83	0	18	59	.521	11.239	.2519	2098.79	3.35	17	22	0	0	3.350	1.000	1.0000	0
	86	0	11	34	.485	11.354	.2511	2122.32		20	0	3	44	3.135	1.365	.8014	.5064
	89	0	2	59	.468	11.407	.2507	2132.92		23	0	7	10	2.825	2.182	.6531	38.510
	90	0	0	0	.467	11.410	.2506	2133.42		26	0	10	54	2.735	2.349	.5531	107.19
	3.20	18	13	0	3.200	1.000	1.0000	0	29	0	14	1	2	2.559	2.911	.4026	207.54
	20	0	2	34	3.059	1.055	.8623	1.359		32	0	16	55	2.381	3.510	.4311	333.09
	23	0	6	27	2.843	1.657	.6997	22.418		35	0	19	38	2.210	4.141	.3924	476.78
	26	0	9	57	2.670	2.129	.5901	74.520		38	0	22	11	2.081	4.796	.3626	632.66
	29	0	13	8	2.500	2.641	.52129	156.77		41	0	24	34	1.935	5.469	.3392	795.35
	32	0	16	5	2.339	3.188	.4564	263.75		44	0	26	48	1.796	6.151	.3206	960.51
	35	0	18	50	2.187	3.764	.4140	389.52		47	0	28	52	1.661	6.836	.3055	1124.67
	38	0	21	24	2.040	4.362	.3814	528.97		50	0	30	44	1.536	7.517	.2932	1285.01
	41	0	23	47	1.900	4.975	.3557	675.74		53	0	32	25	1.415	8.184	.2831	1439.07
	44	0	26	4	1.766	5.598	.3353	826.64		56	0	33	50	1.298	8.832	.2747	1585.62
	47	0	28	7	1.637	6.223	.3188	977.62		59	0	34	59	1.187	9.453	.2677	1722.56
	50	0	30	0	1.514	6.844	.3053	1126.41		62	0	35	47	1.081	10.041	.2619	1849.43
	53	0	31	40	1.395	7.453	.2943	1270.16		65	0	36	8	.980	10.588	.2571	1964.84
	56	0	33	5	1.283	8.045	.2851	1407.01		68	0	35	56	.884	11.089	.2530	2066.30
	59	0	34	13	1.174	8.611	.2774	1535.69		71	0	35	3	.794	11.539	.2497	2159.53
	62	0	34	59	1.071	9.147	.2710	1655.19		74	0	33	15	.710	11.931	.2470	2237.90
	65	0	35	19	.972	9.647	.2657	1764.42		77	0	30	19	.634	12.264	.2449	2303.50
	68	0	35	6	.879	10.104	.2613	1862.53		80	0	25	56	.568	12.531	.2432	2355.54
	71	0	34	11	.791	10.514	.2577	1949.11		83	0	19	54	.514	12.732	.2420	2394.22
	74	0	32	23	.709	10.873	.2547	2023.80		86	0	12	10	.491	12.862	.2413	2419.05
	77	0	29	27	.635	11.176	.2524	2085.90		89	0	3	9	.458	12.922	.2409	2430.52
	80	0	25	8	.573	11.420	.2506	2135.45		90	0	0	0	.457	12.926	.24	

TABLE I - Continued

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,
 AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Continued

M_1	ϵ				δ				M_2	P_2	P_1	$\frac{\rho_2}{\rho_1}$	$\frac{\Delta S}{ft^2/sec^2}$	M_1	ϵ				δ				M_2	P_2	P_1	$\frac{\rho_2}{\rho_1}$	$\frac{\Delta S}{ft^2/sec^2}$
	(deg)	(min)	(deg)	(min)	(deg)	(min)	(deg)	(min)							(deg)	(min)	(deg)	(min)	(deg)	(min)	(deg)	(min)					
3.40	48	0	29	44	1.628	7.282	.2972	1229.91	3.55	62	0	36	41	1.094	11.296	0.2515	2110.35								2233.32		
	51	0	31	33	1.502	7.979	.2860	1391.11		65	0	37	5	.990	11.911	.2472									2343.96		
	54	0	33	9	1.381	8.661	.2768	1547.13		68	0	36	55	.891	12.473	.2436									2441.09		
	57	0	34	29	1.262	9.320	.2692	1694.75		71	0	36	3	.798	12.978	.2406									2524.31		
	60	0	35	32	1.155	9.949	.2628	1829.58		74	0	34	17	.712	13.420	.2382									2588.67		
	63	0	36	12	1.050	10.540	.2575	1954.93		77	0	31	20	.633	13.793	.2363									2638.79		
	66	0	36	23	.950	11.089	.2530	2068.30		80	0	26	54	.564	14.093	.2348									2689.23		
	69	0	35	59	.855	11.588	.2494	2169.44		83	0	20	42	.509	14.318	.2338									2715.95		
	72	0	34	50	.766	12.032	.2464	2258.45		86	0	12	42	.469	14.465	.2331									2727.85		
	75	0	32	4	.684	12.417	.2439	2333.35		89	0	3	18	.450	14.532	.2328									2728.51		
3.45	78	0	29	18	.610	12.731	.2420	2395.19	3.60	90	0	0	0	.449	14.536	.2328											
	81	0	24	22	.547	12.990	.2406	2443.70		16	8	0	0	3.600	1.000	1.0000	0										
	84	0	17	41	.498	13.173	.2396	2476.08		17	0	1	19	3.465	1.126	.9189	.0356										
	87	0	9	22	.466	13.283	.2390	2498.82		20	0	5	23	3.263	1.602	.7163	18.007										
	90	0	0	0	.455	13.320	.2388	2505.96		23	0	8	59	3.044	2.142	.5878	76.282										
	16	51	0	0	3.450	1.000	1.0000	0		26	0	12	16	2.843	2.739	.5013	174.56										
	18	0	1	42	3.346	1.159	.8999	.4693		29	0	15	17	2.654	3.387	.4402	306.06										
	21	0	5	43	3.113	1.617	.7118	19.243		32	0	18	7	2.476	4.079	.3956	462.38										
	24	0	9	17	2.907	2.131	.5899	74.797		35	0	20	46	2.305	4.808	.3621	635.32										
	27	0	12	32	2.717	2.695	.5064	166.63		38	0	23	17	2.144	5.565	.3363	818.37										
3.50	30	0	15	32	2.539	3.305	.4467	288.56	3.65	41	0	25	38	1.990	6.341	.3161	1006.11										
	33	0	18	21	2.371	3.952	.4027	432.93		44	0	27	51	1.844	7.130	.2999	1193.94										
	36	0	20	59	2.210	4.631	.3693	593.00		47	0	29	54	1.704	7.921	.2869	1378.49										
	39	0	23	27	2.056	5.333	.3435	762.36		50	0	31	47	1.574	8.706	.2762	1557.16										
	42	0	25	47	1.910	6.051	.3230	936.17		53	0	33	28	1.444	9.478	.2674	1727.56										
	45	0	27	57	1.770	6.777	.3067	1111.59		56	0	34	54	1.323	10.226	.2602	1888.70										
	48	0	29	57	1.636	7.502	.2934	1281.49		59	0	36	4	1.207	10.943	.2542	2038.24										
	51	0	31	46	1.504	8.220	.2826	1447.34		62	0	36	54	1.097	11.621	.2489	2169.45										
	54	0	33	22	1.387	8.922	.2736	1605.36		65	0	37	17	.992	12.253	.2449	2301.30										
	57	0	34	43	1.270	9.601	.2662	1754.71		68	0	37	8	.893	12.832	.2415	2413.29										
3.55	60	0	35	45	1.159	10.488	.2600	1893.51	3.70	71	0	36	17	.799	13.351	.2386	2511.45										
	63	0	36	26	1.053	10.858	.2549	2020.94		74	0	34	31	.633	14.189	.2344	2665.81										
	66	0	36	38	.952	11.422	.2506	2136.31		77	0	31	34	.564	14.498	.2330	2721.83										
	69	0	36	14	.856	11.936	.2470	2239.14		80	0	27	7	.564	14.498	.2330	2721.83										
	72	0	35	5	.767	12.394	.2441	2282.84		83	0	20	54	.507	14.729	.2319	2762.93										
	75	0	32	57	.684	12.790	.2417	2405.02		86	0	12	49	.468	14.880	.2313	2787.74										
	78	0	29	33	.610	13.119	.2398	2467.98		89	0	3	19	.449	14.949	.2310	2801.83										
	81	0	24	36	.546	13.380	.2384	2497.27		90	0	0	0	.447	14.953	.2310	2802.74										
	84	0	17	53	.496	13.568	.2375	2524.10																			
	87	0	9	22	.466	13.619	.2369	2573.13																			
	90	0	0	0	.453	13.720	.2367	2580.25																			
3.60	16	36	0	0	3.500	1.000	1.0000	0	3.70	15	54	0	0	3.650	1.000	1.0000	0										
	17	0	0	36	3.462	1.055	.9625	-----		18	0	3	3	3.453	1.318	.8017	3.597										
	20	0	4	46	3.209	1.505	.7482	11.778		21	0	6	55	3.213	1.830	.6537	38.249										
	23	0	8	26	2.997	2.015	.6122	59.476		24	0	10	22	2.998	2.405	.5448	116.07										
	26	0	11	45	2.800	2.580	.5206	145.61		27	0	13	32	2.799	3.037	.4702	232.78										
	29	0	14	49	2.617	3.193	.4561	264.53		30	0	16	28	2.611	3.719	.4169	379.76										
	32	0	17	40	2.443	3.847	.4089	418.39		33	0	19	42	2.434	4.444	.3775	548.35										
	35	0	20	21	2.277	4.535	.3734	570.04		36	0	21	49	2.264	5.203	.3477	731.01										
	38	0	22	52	2.131	5.251	.3461	742.39		39	0	24	50	2.103	5.989	.3246	921.31										
	41	0	25	14	1.970	5.985	.3247	920.04		42	0	26	55	1.951	6.792	.3064	1114.11										
3.70	44	0	27	27	1.819	6.730	.3076	1099.16		45	0	28	44	1.805	7.605	.2918	1305.54										
	47	0	30	30	1.688	7.478	.2938	1275.69		48	0	30	44	1.666	8.417	.2799	1492.14										
	50	0	31	23	1.557	8.220	.2826	1447.18		49	0	33	51	1.520	9.392	.2558	1692.72										
	53	0	33	7	1.433	8.949	.2733	1611.78		50	0	35	52	1.434	10.355	.2337	1901.50										
	56	0	34	30	1.313	9.656	.2656	1766.51		51	0	36	53	1.327	11.359	.2337	2150.14										
	59	0	35	40	1.091	10.975	.2539	2042.6																			

TABLE I - Concluded

VALUES OF DEVIATION OF FLOW, MACH NUMBER, PRESSURE RATIO, DENSITY RATIO,

AND CHANGE IN ENTROPY ACROSS SHOCK WAVES - Concluded

M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec^2) \frac{OF}{OF}}$	M ₁	ε		δ		M ₂	$\frac{P_2}{P_1}$	$\frac{\rho_1}{\rho_2}$	$\frac{\Delta S}{(ft^2/sec^2) \frac{OF}{OF}}$
	(deg)	(min)	(deg)	(min)						(deg)	(min)	(deg)	(min)				
3.70	75	0	34	7	0.685	14.735	0.2319	2764.36	3.85	89	0	3	29	0.440	17.122	0.2229	3170.36
	78	0	30	43	.668	15.115	.2303	2831.36		90	0	0	29	.439	17.126	.2229	3171.00
	81	0	25	40	.542	15.414	.2291	2883.64									
	84	0	18	44	.490	15.630	.2282	2920.32									
	87	0	9	58	.456	15.761	.2277	2943.39									
	90	0	0	0	.444	15.805	.2275	2950.81									
3.75	15	28	0	0	3.750	1.000	1.0000	0		29	0	0	29	0.440	17.122	0.2229	3170.36
	18	0	3	39	3.506	1.400	.7873	6.776		30	0	0	30	.761	4.004	.3998	444.78
	21	0	7	27	3.263	1.940	.6281	50.411		31	0	0	31	2.570	4.817	.3618	639.36
	24	0	10	51	3.043	2.548	.5249	140.21		32	0	0	32	5.671	3.332	.844.31	
	27	0	13	58	2.838	3.215	.4542	269.45		33	0	0	33	3.418	1.907	.6350	40.704
	30	0	16	53	2.646	3.935	.4037	428.93		34	0	0	34	2.542	.5255	139.05	
	33	0	19	37	2.464	4.700	.3664	609.50		35	0	0	35	1.985	8.396	.2802	1187.37
	36	0	22	12	2.291	5.502	.3392	803.26		36	0	0	36	1.747	9.325	.2691	1694.37
	39	0	24	38	2.126	6.331	.3163	1003.82		37	0	0	37	10.247	.2600	1897.88	
	42	0	26	56	1.970	7.179	.2990	1205.69		38	0	0	38	1.475	11.152	.2526	2081.24
	45	0	29	5	1.821	8.037	.2852	1391.39		39	0	0	39	1.348	12.030	.2464	2257.39
	48	0	31	5	1.680	8.894	.2740	1599.23		40	0	0	40	7.471	.2940	2124.62	
	51	0	32	54	1.546	9.742	.2648	1785.17		41	0	0	41	1.114	13.668	.2369	2570.33
	54	0	34	30	1.418	10.571	.2572	1961.46		42	0	0	42	1.005	14.409	.2334	2705.87
	57	0	35	52	1.296	11.373	.2509	2126.34		43	0	0	43	.902	15.088	.2304	2826.46
	60	0	37	6	1.186	12.138	.2457	2278.71		44	0	0	44	.805	15.698	.2279	2932.18
	63	0	37	40	1.070	12.858	.2413	2418.46		45	0	0	45	.715	16.230	.2260	3008.81
	66	0	37	54	.964	13.526	.2377	2544.18		46	0	0	46	.632	16.681	.2244	3097.97
	69	0	37	34	.865	14.133	.2347	2656.08		47	0	0	47	.560	17.044	.2232	3157.53
	72	0	36	27	.772	14.673	.2322	2753.23		48	0	0	48	.501	17.316	.2223	3201.68
	75	0	34	20	.665	15.141	.2302	2835.86		49	0	0	49	.459	17.492	.2217	3230.39
	78	0	30	55	.608	15.531	.2286	2903.55		50	0	0	50	.439	17.573	.2215	3243.46
	81	0	25	51	.541	15.838	.2274	2956.35		51	0	0	51	.438	17.578	.2215	3244.11
	84	0	18	53	.488	16.060	.2266	2994.16		52	0	0	52				
	87	0	10	3	.455	16.195	.2261	3016.89		53	0	0	53				
	90	0	0	0	.442	16.240	.2259	3024.28		54	0	0	54				
3.80	15	15	0	0	3.800	1.000	1.0000	0	3.95	14	0	0	0	3.950	1.000	1.0000	0
	17	0	2	34	3.624	1.273	.8418	2.328		15	0	0	24	3.707	1.389	.7914	5.742
	20	0	6	30	3.368	1.804	.6600	35.512		16	0	0	25	3.441	1.963	.6232	52.900
	23	0	9	59	3.137	2.405	.5447	116.10		17	0	0	26	3.205	2.613	.5165	151.47
	26	0	13	10	2.925	3.071	.4670	239.51		18	0	0	27	2.985	3.332	.4446	294.11
	29	0	16	8	2.727	3.793	.4122	306.33		19	0	0	28	2.778	4.112	.3939	469.90
	32	0	18	54	2.538	4.564	.3722	576.93		20	0	0	29	2.584	4.945	.3568	668.50
	35	0	21	32	2.361	5.376	.3421	772.71		21	0	0	30	2.399	5.822	.3290	880.96
	38	0	24	1	2.192	6.219	.3189	976.54		22	0	0	31	2.059	7.669	.2908	1320.30
	41	0	26	21	2.031	7.085	.3007	1183.35		23	0	0	32	1.902	8.618	.2773	1537.36
	44	0	28	33	1.879	7.963	.2863	1384.22		24	0	0	33	1.774	9.570	.2665	1747.86
	47	0	30	36	1.734	8.844	.2746	1588.04		25	0	0	34	1.614	10.516	.2577	1949.57
	50	0	32	28	1.596	9.720	.2650	1780.24		26	0	0	35	1.480	11.444	.2504	2140.49
	53	0	34	10	1.465	10.579	.2571	1962.64		27	0	0	36	1.353	12.345	.2444	2319.36
	56	0	35	37	1.340	11.401	.2506	2129.67		28	0	0	37	1.208	13.208	.2394	2484.70
	59	0	36	48	1.222	12.212	.2452	2292.69		29	0	0	38	1.116	14.025	.2352	2636.29
	62	0	37	39	1.136	12.967	.2407	2438.95		30	0	0	39	1.007	14.786	.2317	2773.19
	65	0	38	4	1.001	13.672	.2369	2571.26		31	0	0	40	.903	15.483	.2288	2895.03
	68	0	37	57	.899	14.316	.2338	2689.05		32	0	0	41	.806	16.108	.2264	2992.18
	71	0	37	7	.803	14.895	.2312	2792.42		33	0	0	42	.700	17.767	.2209	3229.18
	74	0	35	22	.714	15.400	.2291	2880.82		34	0	0	43	.600	17.489	.2217	3274.19
	77	0	32	26	.632	15.828	.2274	2954.38		35	0	0	44	.508	17.949	.2203	3302.83
	80	0	27	56	.561	16.172	.2262	3013.13		36	0	0	45	.438	18.032	.2201	3316.12
	83	0	21	36	.503	16.430	.2252	3056.29		37	0	0	46	.436	18.036	.2201	3316.92
	86	0	13	17	.462	16.598	.2247	3084.18		38	0	0	47				
	89	0	3	27	.442	16.675	.2244	3096.82		39	0	0	48				
	90	0	0	0	.441	16.680	.2244	3097.73		40	0	0	49				
3.85	15	3	0	0	3.850	1.000	1.0000	0	4.00	14	29	0	0	4.000	1.000	1.0000	0
	17	0	2	45	3.655	1.312	.8243	3.462		15	0	0	30	3.467	2.017	.6119	59.660
	20	0	6	45	3.392	1.856	.6472	40.898		16	0	0	31	3.004	3.421	.4377	313.26
	23	0	10	12	3.159	2.474	.5349	127.38		17	0	0	32	2.795	4.221	.3882	495.42
	26	0	13	23	2.946	3.157	.4592	243.39		18	0	0	33	2.598	5.075	.3521	699.85
	29	0	16	19	2.734	3.898	.4059	420.41		19	0	0	34	2.335	6.909	.3041	1141.61
	32	0	19	42	2.373	5.523	.3375	808.22		20	0	0	35	2.069	7.868	.2877	1366.26
	35	0	21	42	2.203	6.388	.3150	1017.35		21	0	0	36	1.911	8.841	.2746	1587.36
	38	0	24	11	2.023	7.277	.2973	1228.64		22	0	0	37	1.768	9.818	.2640	1801.39
	41	0	26	31	2.041	7.277	.2973	1423.73		23	0	0	38	1.619	10.788	.2554	2006.14
	44	0	28	42	1.886	8.178	.2832	1640.95		24	0	0	39	1.485	11.740	.2479	2190.53
	47	0	30	45	1.741	9.083	.2718	1836.43		25	0	0	40	1.357	12.664	.2424	2380.55
	50	0	32	36	1.602												

TABLE II

ANGLE OF SHOCK AND M_{2m} FOR MAXIMUM DEVIATION OF THE FLOW
 AND ANGLE OF SHOCK AND δ , WHICH GIVE
 SONIC VELOCITY BEHIND THE SHOCK

M_1	ϵ_m		ϵ_s		δ_m		δ_s		M_{2m}
	(deg)	(min)	(deg)	(min)	(deg)	(min)	(deg)	(min)	
1.0	90		90		0		0		1.000
1.1	76	18	73	14	1	31	1	24	.9710
1.2	71	59	68	5	3	57	3	42	.9500
1.3	69	24	65	7	6	40	6	19	.9357
1.4	67	42	63	20	9	26	9	1	.9268
1.5	66	36	62	15	12	6	11	41	.9212
1.6	65	50	61	39	14	39	14	15	.9187
1.7	65	19	61	22	17	0	16	38	.9185
1.8	64	59	61	17	19	11	18	50	.9196
1.9	64	47	61	21	21	10	20	52	.9216
2.0	64	40	61	29	22	59	22	43	.9243
2.1	64	37	61	41	24	37	24	23	.9274
2.2	64	37	61	54	26	6	25	54	.9306
2.3	64	41	62	9	27	28	27	17	.9331
2.4	64	42	62	24	28	42	28	32	.9374
2.5	64	48	62	39	29	48	29	40	.9397
2.6	64	52	62	53	30	49	30	42	.9426
2.7	64	57	63	7	31	45	31	39	.9464
2.8	65	3	63	21	32	35	32	30	.9489
2.9	65	9	63	33	33	21	33	17	.9514
3.0	65	15	63	46	34	4	34	1	.9537
3.1	65	20	63	58	34	44	34	40	.9565
3.2	65	25	64	8	35	20	35	17	.9589
3.3	65	31	64	18	35	53	35	51	.9606
3.4	65	36	64	28	36	24	36	22	.9627
3.5	65	41	64	37	36	52	36	50	.9645
3.6	65	46	64	45	37	18	37	17	.9660
3.7	65	51	64	54	37	43	37	41	.9674
3.8	65	56	65	2	38	5	38	4	.9685
3.9	65	59	65	9	38	27	38	25	.9708
4.0	66	3	65	15	38	47	38	45	.9721



TABLE III

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
AND MACH NUMBER BEHIND THE SHOCK

M_1	δ		ϵ		M_2	M_1	δ		ϵ		M_2
	(deg)	(min)	(deg)	(min)			(deg)	(min)	(deg)	(min)	
1.05	0	0	72	15	1.050	1.40	5	0	52	46	1.216
	0	30	78	0	1.000		5	30	53	39	1.195
1.10	0	0	65	23	1.100	1.45	6	0	54	38	1.174
	0	30	67	20	1.074		6	30	55	40	1.151
1.15	0	0	69	54	1.041	1.45	7	0	56	46	1.127
	1	30	74	40	.982		7	30	57	58	1.102
1.20	0	0	60	24	1.150	1.45	8	0	59	22	1.074
	0	30	61	42	1.127		9	30	61	6	1.042
1.25	0	0	63	11	1.104	1.50	0	0	43	36	1.450
	1	30	64	54	1.075		0	30	44	11	1.433
1.30	0	0	66	58	1.044	1.50	1	0	44	48	1.416
	1	30	70	38	.998		1	30	45	24	1.397
1.35	0	0	56	26	1.200	1.50	2	0	46	40	1.362
	0	30	57	27	1.180		2	30	47	10	1.344
1.40	0	0	58	54	1.158	1.55	3	0	48	1	1.326
	1	30	59	45	1.135		3	30	48	42	1.307
1.45	0	0	61	3	1.110	1.55	4	0	48	24	1.290
	1	30	62	32	1.083		4	30	49	7	1.271
1.50	0	0	64	25	1.053	1.55	5	0	50	55	1.252
	0	30	66	56	1.019		5	30	50	45	1.232
1.55	0	0	53	8	1.250	1.55	6	0	52	36	1.211
	0	30	53	58	1.232		7	0	53	30	1.190
1.60	0	0	54	54	1.212	1.55	7	30	54	28	1.168
	1	30	55	51	1.192		8	0	55	30	1.146
1.65	0	0	56	52	1.171	1.55	8	30	56	38	1.120
	1	30	57	58	1.149		9	0	57	54	1.095
1.70	0	0	59	10	1.126	1.55	9	30	59	19	1.067
	1	30	60	29	1.100		10	0	61	5	1.036
1.75	0	0	62	5	1.072	1.55	10	30	63	38	.987
	0	30	63	59	1.040		0	0	41	49	1.500
1.80	0	0	66	32	.998	1.55	0	30	42	22	1.487
	1	30	66	32	.998		1	0	42	55	1.466
1.85	0	0	50	17	1.300	1.55	1	30	43	30	1.449
	0	30	51	2	1.282		2	0	44	5	1.431
1.90	0	0	51	50	1.264	1.55	2	30	44	42	1.414
	1	30	52	39	1.245		3	0	45	19	1.396
1.95	0	0	53	30	1.225	1.55	3	30	45	55	1.380
	1	30	54	24	1.206		4	0	46	34	1.362
2.00	0	0	55	11	1.186	1.55	4	30	47	14	1.344
	1	30	56	20	1.165		5	0	47	54	1.326
2.05	0	0	57	26	1.142	1.55	5	30	48	37	1.309
	1	30	58	36	1.117		6	0	49	22	1.290
2.10	0	0	59	58	1.091	1.55	6	30	50	6	1.271
	1	30	61	32	1.062		7	0	50	54	1.251
2.15	0	0	63	39	1.028	1.55	7	30	51	44	1.230
	1	30	66	28	.980		8	0	52	34	1.209
2.20	0	0	47	47	1.350	1.55	8	30	53	30	1.187
	0	30	48	29	1.332		9	0	54	26	1.164
2.25	0	0	49	10	1.315	1.55	9	30	55	30	1.139
	1	30	49	55	1.297		10	0	56	42	1.113
2.30	0	0	50	41	1.278	1.55	10	30	57	58	1.086
	1	30	51	18	1.259		11	0	59	29	1.055
2.35	0	0	52	16	1.240	1.55	11	30	61	25	1.016
	1	30	53	6	1.219		12	0	64	24	.956
2.40	0	0	53	59	1.199	1.55	0	0	40	11	1.550
	0	30	54	56	1.177		0	30	40	41	1.532
2.45	0	0	55	57	1.156	1.55	1	0	41	14	1.515
	1	30	57	3	1.131		1	30	41	47	1.497
2.50	0	0	58	17	1.106	1.55	2	0	42	10	1.480
	1	30	59	37	1.080		2	30	42	54	1.463
2.55	0	0	61	14	1.050	1.55	3	0	43	29	1.446
	1	30	63	15	1.017		3	30	44	4	1.428
2.60	0	0	66	58	.952	1.55	4	0	44	41	1.412
	1	30	66	58	.952		4	30	45	18	1.395
2.65	0	0	45	35	1.400	1.55	5	0	45	55	1.376
	0	30	46	10	1.382		5	30	46	34	1.360
2.70	0	0	46	49	1.365	1.55	6	0	47	14	1.341
	1	30	47	28	1.347		6	30	47	55	1.324
2.75	0	0	48	9	1.330	1.55	7	0	48	40	1.306
	1	30	48	52	1.311		7	30	49	24	1.286
2.80	0	0	49	35	1.293	1.55	8	0	50	10	1.267
	1	30	50	19	1.275		8	30	50	59	1.246
2.85	0	0	51	6	1.256	1.55	9	0	51	50	1.225
	1	30	51	55	1.236		9	30	52	41	1.204



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
 AND MACH NUMBER BEHIND THE SHOCK - Continued

M ₁	δ		ε		M ₂	M ₁	δ		ε		M ₂
	(deg)	(min)	(deg)	(min)			(deg)	(min)	(deg)	(min)	
1.55	10	0	53	17	1.181	1.70	3	30	39	26	1.580
	10	30	54	36	1.156		4	0	39	58	1.563
	11	0	55	38	1.131		4	30	40	30	1.546
	11	30	56	52	1.105		5	0	41	2	1.528
	12	0	58	14	1.075		5	30	41	35	1.511
	12	30	59	49	1.040		6	0	42	9	1.494
	13	0	62	0	1.000		7	30	42	18	1.458
									43		
									54		
									30		
									44		
									30		
									45		
									10		
									49		
1.60	0	0	38	41	1.600	1.75	7	30	47	29	1.440
	0	30	39	14	1.582		8	0	47	10	1.421
	1	0	39	47	1.565		8	30	48	37	1.403
	1	30	40	19	1.548		9	0	49	22	1.379
	2	0	40	53	1.531		9	30	50	9	1.365
	2	30	41	27	1.515		10	0	51	0	1.346
	3	0	42	1	1.497		10	30	52	53	1.327
	3	30	42	34	1.480		11	0	53	47	1.307
	4	0	43	10	1.462		11	30	54	52	1.287
	4	30	43	45	1.445		12	0	55	0	1.267
	5	0	44	19	1.427		12	30	56	1.245	
	5	30	44	55	1.410		13	0	57	53	
	6	0	45	32	1.391		13	30	58	47	
	6	30	46	10	1.374		14	0	59	48	
	7	0	46	47	1.356		14	30	60	52	
	7	30	47	26	1.338		15	0	61	0	
1.65	8	0	48	7	1.320	1.75	15	30	62	40	1.091
	8	30	48	50	1.300		16	0	63	40	1.057
	9	0	49	32	1.281		16	30	64	40	1.014
	9	30	50	14	1.261		17	0	64	40	.930
	10	0	51	6	1.240						
	10	30	51	58	1.218						
	11	0	52	53	1.196						
	11	30	53	53	1.174						
	12	0	54	56	1.148						
	12	30	56	1	1.122						
	13	0	57	14	1.093						
	13	30	58	44	1.062						
	14	0	60	37	1.024						
	14	30	63	20	.965						
1.70						1.80					



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
 AND MACH NUMBER BEHIND THE SHOCK - Continued

M ₁	δ		ε		M ₂	M ₁	δ		ε		M ₂
	(deg)	(min)	(deg)	(min)			(deg)	(min)	(deg)	(min)	
1.80	6	30	40	2	1.575	1.90	5	0	36	13	1.725
	7	0	40	34	1.557		5	30	36	42	1.707
	7	30	41	8	1.540		6	0	37	13	1.690
	8	0	41	42	1.523		6	30	37	43	1.672
	8	30	42	16	1.505		7	0	38	14	1.654
	9	0	42	50	1.487		7	30	38	45	1.636
	9	30	43	26	1.468		8	0	39	17	1.618
	10	0	44	2	1.450		8	30	39	49	1.600
	10	30	44	42	1.431		9	0	40	21	1.582
	11	0	45	22	1.411		9	30	41	29	1.565
	11	30	46	1	1.392		10	0	42	4	1.547
	12	0	46	41	1.372		11	0	42	40	1.528
	12	30	47	24	1.352		11	30	43	40	1.510
	13	0	48	9	1.332		11	0	43	16	1.490
	13	30	48	54	1.311		12	0	43	52	1.471
	14	0	49	39	1.289		12	30	44	30	1.452
	14	30	50	28	1.267		13	0	45	10	1.433
	15	0	51	20	1.243		13	30	45	51	1.413
	15	30	52	14	1.219		14	0	46	33	1.392
	16	0	53	10	1.195		14	30	47	16	1.371
	16	30	54	15	1.170		15	0	48	0	1.349
	17	0	55	22	1.142		15	30	48	44	1.327
	17	30	56	36	1.112		16	0	49	32	1.305
	18	0	58	0	1.077		16	30	50	20	1.272
	18	30	59	42	1.035		17	0	51	12	1.259
	19	0	62	18	.972		17	30	52	4	1.234
1.85	0	0	32	43	1.850		18	0	53	2	1.209
	0	30	33	9	1.832		18	30	54	5	1.182
	1	0	33	35	1.816		19	0	55	14	1.152
	1	30	34	2	1.797		19	30	56	29	1.120
	2	0	34	29	1.780		20	0	57	52	1.084
	2	30	34	56	1.764		20	30	59	36	1.043
	3	0	35	23	1.746		21	0	62	6	.981
	3	30	35	50	1.729	1.95	0	0	30	51	1.950
	4	0	36	19	1.711		0	30	31	16	1.932
	4	30	36	47	1.694		1	0	31	42	1.915
	5	0	37	17	1.676		1	30	32	7	1.897
	5	30	37	47	1.659		2	0	32	33	1.879
	6	0	38	18	1.641		2	30	32	59	1.862
	6	30	28	49	1.624		3	0	33	25	1.844
	7	0	39	20	1.607		3	30	33	51	1.826
	7	30	40	12	1.590		4	0	34	19	1.809
	8	0	40	25	1.571		4	30	34	46	1.791
	8	30	40	58	1.553		5	0	35	14	1.773
	9	0	41	34	1.535		5	30	35	42	1.756
	9	30	42	9	1.516		6	0	36	11	1.738
	10	0	42	43	1.498		6	30	36	41	1.720
	10	30	43	19	1.480		7	0	37	10	1.703
	11	0	43	56	1.460		7	30	37	42	1.685
	11	30	44	33	1.441		8	0	38	12	1.666
	12	0	45	12	1.423		8	30	38	44	1.647
	12	30	45	53	1.403		9	0	39	15	1.629
	13	0	46	14	1.384		9	30	39	48	1.611
	13	30	47	18	1.364		10	0	40	20	1.593
	14	0	48	2	1.344		10	30	40	54	1.575
	14	30	48	46	1.322		11	0	41	29	1.556
	15	0	49	33	1.300		11	30	42	5	1.537
	15	30	50	22	1.277		12	0	42	40	1.519
	16	0	51	13	1.254		12	30	43	16	1.500
	16	30	52	8	1.230		13	0	43	55	1.480
	17	0	53	6	1.204		13	30	44	34	1.459
	17	30	54	6	1.177		14	0	45	14	1.439
	18	0	55	14	1.147		14	30	45	55	1.418
	18	30	56	26	1.117		15	0	46	36	1.397
	19	0	57	48	1.083		15	30	47	20	1.376
	19	30	59	28	1.042		16	0	48	4	1.355
	20	0	62	14	.983		16	30	48	50	1.333
							17	0	49	37	1.310
1.90	0	0	31	45	1.900		17	30	50	26	1.286
	0	30	32	9	1.884		18	0	51	19	1.261
	1	0	32	35	1.866		18	30	52	14	1.236
	1	30	33	1	1.847		19	0	53	10	1.210
	2	0	33	28	1.830		19	30	54	15	1.188
	2	30	33	54	1.813		20	0	55	22	1.152
	3	0	34	22	1.795		20	30	56	38	1.121
	3	30	34	48	1.777		21	0	58	6	1.084
	4	0	35	16	1.760		21	30	59	46	1.038
	4	30	35	44	1.743		22	0	62	10	.963



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,

AND MACH NUMBER BEHIND THE SHOCK - Continued

M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
2.00	0	30	0	2.000	2.10	18	47	13	1.407	2.25	4	29	32	2.096
	1	30	50	1.964		19	48	44	1.361		5	30	24	2.059
	2	31	40	1.927		20	50	22	1.312		6	31	16	2.021
	3	32	30	1.892		21	52	7	1.260		7	32	10	1.982
	4	33	23	1.856		22	54	9	1.202		8	33	5	1.944
	5	34	19	1.821		23	56	34	1.136		9	34	2	1.905
	6	35	16	1.785		24	59	48	1.051		10	35	2	1.866
	7	36	13	1.749							11	36	1	1.827
	8	37	14	1.713	2.15	0	27	43	2.150		12	37	4	1.788
	9	38	16	1.676		1	28	30	2.112		13	38	8	1.747
	10	39	19	1.639		2	29	19	2.075		14	39	14	1.706
	11	40	25	1.602		3	30	7	2.039		15	40	24	1.666
	12	41	33	1.565		4	30	57	2.002		16	41	36	1.624
	13	42	44	1.526		5	31	48	1.964		17	42	50	1.581
	14	44	1	1.486		6	32	42	1.927		18	44	8	1.538
	15	45	20	1.447		7	33	37	1.890		19	45	29	1.495
	16	46	42	1.405		8	34	34	1.852		20	46	56	1.447
	17	48	10	1.360		9	35	33	1.814		21	48	27	1.399
	18	49	42	1.313		10	36	34	1.776		22	50	4	1.347
	19	51	9	1.264		11	37	38	1.739		23	51	50	1.292
	20	53	20	1.211		12	38	43	1.700		24	53	49	1.233
	21	55	32	1.151		13	39	49	1.662		25	56	8	1.164
	22	58	24	1.077		14	40	58	1.620		26	59	8	1.080
2.05	0	29	12	2.050		15	42	10	1.579	2.30	0	25	46	2.300
	1	30	0	2.014		16	43	25	1.537		1	26	30	2.260
	2	30	50	1.978		17	44	43	1.495		2	27	16	2.222
	3	31	40	1.943		18	46	5	1.451		3	28	4	2.183
	4	32	32	1.907		19	47	33	1.406		4	28	54	2.144
	5	33	23	1.871		20	49	8	1.358		5	29	44	2.105
	6	34	22	1.834		21	50	48	1.307		6	30	35	2.067
	7	35	18	1.797		22	52	36	1.252		7	31	30	2.028
	8	36	16	1.760		23	54	42	1.192		8	32	25	1.989
	9	37	18	1.723		24	57	14	1.122		9	33	21	1.949
	10	38	22	1.685		25	60	52	1.033		10	34	19	1.910
	11	39	26	1.648	2.20	0	27	2	2.200		11	35	20	1.870
	12	40	32	1.609		1	27	48	2.161		12	36	22	1.831
	13	41	42	1.570		2	28	35	2.123		13	37	25	1.791
	14	42	54	1.531		3	29	24	2.086		14	38	30	1.750
	15	44	11	1.491		4	30	14	2.049		15	39	38	1.708
	16	45	30	1.450		5	31	5	2.011		16	40	48	1.667
	17	46	56	1.406		6	31	58	1.973		17	42	0	1.625
	18	48	26	1.361		7	32	52	1.935		18	43	16	1.581
	19	50	2	1.314		8	33	50	1.897		19	44	36	1.535
	20	51	46	1.263		9	34	49	1.860		20	46	0	1.489
	21	53	42	1.209		10	35	48	1.822		21	47	26	1.441
	22	56	2	1.145		11	36	50	1.784		22	49	2	1.390
	23	59	6	1.065		12	37	52	1.745		23	50	38	1.336
2.10	0	28	26	2.100		13	38	58	1.705		24	52	30	1.280
	1	29	13	2.064		14	40	4	1.664		25	54	29	1.217
	2	30	2	2.027		15	41	14	1.623		26	57	2	1.142
	3	30	52	1.991		16	42	28	1.582		27	60	38	1.046
	4	31	44	1.955		17	43	44	1.539					
	5	32	37	1.918		18	45	4	1.494	2.35	0	25	11	2.350
	6	33	41	1.882		19	46	28	1.449		1	25	55	2.311
	7	34	28	1.846		20	47	58	1.403		2	26	42	2.272
	8	35	25	1.809		21	49	32	1.354		3	27	28	2.234
	9	36	24	1.771		22	51	14	1.301		4	28	18	2.195
	10	37	26	1.732		23	53	8	1.246		5	29	8	2.155
	11	38	30	1.695		24	55	18	1.184		6	29	58	2.116
	12	39	36	1.656		25	58	2	1.107		7	30	51	2.076
	13	40	42	1.617		26	62	56	.982		8	31	46	2.037
	14	41	54	1.577	2.25	0	26	23	2.250		9	32	42	1.997
	15	43	10	1.537		1	27	6	2.212		11	34	38	1.917
	16	44	28	1.495		2	27	54	2.173		12	35	39	1.876
	17	45	49	1.451		3	28	42	2.135		13	36	42	1.834

TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
 AND MACH NUMBER BEHIND THE SHOCK - Continued

M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2	M_1	δ (deg)	ϵ		M_2
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
2.35	14	37	46	1.793	2.45	21	44	57	1.560	2.60	26	50	17	1.401
	15	38	54	1.751		22	46	21	1.510		27	52	6	1.340
	16	40	2	1.709		23	47	50	1.459		28	54	4	1.272
	17	41	15	1.666		24	49	25	1.405		29	56	22	1.197
	18	42	28	1.622		25	51	7	1.349		30	59	22	1.105
	19	43	45	1.576		26	53	1	1.287					
	20	45	6	1.530		27	55	12	1.219		0	21	44	2.700
	21	46	31	1.483		28	57	52	1.140		1	22	26	2.652
	22	48	2	1.432		29	61	51	1.023		2	23	10	2.608
	23	49	38	1.380							3	23	56	2.564
	24	51	23	1.324	2.50	0	23	35	2.500		4	24	43	2.519
	25	53	16	1.264		1	24	19	2.459		5	25	30	2.474
	26	55	30	1.196		2	25	5	2.418		6	26	19	2.430
	27	58	14	1.116		3	25	52	2.376		7	27	10	2.386
	28	63	0	.980		4	26	38	2.335		8	28	2	2.343
2.40	0	24	37	2.400		5	27	26	2.294		9	28	55	2.300
	1	25	21	2.360		6	28	17	2.252		10	29	50	2.256
	2	26	7	2.320		7	29	8	2.210		11	30	45	2.211
	3	26	54	2.280		8	30	1	2.169		12	31	42	2.166
	4	27	42	2.241		9	30	56	2.126		13	32	42	2.121
	5	28	32	2.201		10	31	52	2.085		14	33	44	2.075
	6	29	22	2.160		11	32	49	2.038		15	34	47	2.029
	7	30	14	2.120		12	33	47	2.000		16	35	52	1.982
	8	31	8	2.080		13	34	48	1.957		17	36	58	1.935
	9	32	4	2.041		14	35	52	1.916		18	38	6	1.887
	10	33	0	2.000		15	36	56	1.871		19	39	17	1.839
	11	33	58	1.959		16	38	5	1.827		20	40	30	1.790
	12	34	59	1.917		17	39	14	1.783		21	41	45	1.740
	13	36	0	1.876		18	40	24	1.739		22	43	3	1.690
	14	37	6	1.835		19	41	37	1.692		23	44	23	1.638
	15	38	13	1.792		20	42	54	1.645		24	45	48	1.584
	16	39	22	1.748		21	44	14	1.598		25	47	19	1.530
	17	40	32	1.704		22	45	35	1.548		26	48	52	1.472
	18	41	46	1.660		23	47	2	1.496		27	50	32	1.412
	19	43	2	1.615		24	48	38	1.443		28	52	20	1.345
	20	44	20	1.570		25	50	18	1.387		29	54	19	1.280
	21	45	43	1.522		26	52	5	1.327		30	56	36	1.202
	22	47	10	1.472		27	54	0	1.263		31	59	38	1.198
	23	48	44	1.421	2.60	28	56	20	1.189					
	24	50	23	1.367		29	59	17	1.100		0	20	56	2.800
2.45	25	52	10	1.309		0	22	37	2.600		1	21	38	2.753
	26	54	9	1.243		1	23	20	2.556		2	22	22	2.707
	27	56	32	1.169		2	24	6	2.512		3	23	6	2.661
	28	59	45	1.077		3	24	50	2.470		4	23	52	2.615
	0	24	5	2.450		4	25	37	2.426		5	24	38	2.569
	1	24	50	2.408		5	26	26	2.384		6	25	26	2.523
	2	25	34	2.367		6	27	15	2.341		7	27	9	2.476
	3	26	21	2.326		7	28	6	2.299		8	28	2	2.431
	4	27	9	2.286		8	28	58	2.257		9	28	56	2.386
	5	27	57	2.246		9	29	52	2.214		10	28	56	2.339
	6	28	48	2.205		10	30	47	2.173		11	29	52	2.293
	7	29	40	2.163		11	31	45	2.129		12	30	50	2.247
	8	30	33	2.122		12	32	43	2.084		13	31	48	2.200
	9	31	27	2.081		13	33	43	2.039		14	32	48	2.153
	10	32	23	2.041		14	34	45	1.995		15	33	50	2.106
	11	33	22	2.000		15	35	48	1.950		16	34	54	2.057
	12	34	22	1.958		16	36	54	1.906		17	36	0	2.010
	13	35	24	1.917		17	38	2	1.860		18	37	8	1.961
	14	36	28	1.876		18	39	11	1.813		19	38	17	1.912
	15	37	34	1.834		19	40	23	1.767		20	39	30	1.863
	16	38	42	1.790		20	41	38	1.719		21	40	44	1.811
	17	39	50	1.746		21	42	55	1.671		22	42	0	1.757
	18	41	2	1.701		22	44	14	1.621		23	43	20	1.704
	19	42	16	1.655		23	45	38	1.569		24	44	42	1.649
	20	43	36	1.608		24	47	6	1.516		25	46	9	1.594
						25	48	38	1.459		26	47	37	1.539
											27	49	12	1.480

TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
 AND MACH NUMBER BEHIND THE SHOCK - Continued

M ₁	δ (deg)	ε		M ₂	M ₁	δ (deg)	ε		M ₂	M ₁	δ (deg)	ε		M ₂
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
2.80	28	50	54	1.418	3.00	26	45	32	1.659	3.10	22	39	28	1.946
	29	52	42	1.350		27	47	0	1.600		23	40	43	1.890
	30	54	48	1.279		28	48	34	1.539		24	42	0	1.832
	31	57	9	1.198		29	50	14	1.475		25	43	20	1.775
	32	60	25	1.093		30	52	0	1.407		26	44	44	1.715
2.90	0	20	10	2.900	3.05	32	56	7	1.254	3.15	28	47	40	1.593
	1	20	52	2.852		0	19	8	3.050		29	49	50	1.528
	2	21	35	2.804		1	19	50	3.000		31	52	46	1.391
	3	22	19	2.755		2	20	33	2.949		32	54	50	1.316
	4	23	4	2.709		3	21	18	2.898		33	57	10	1.230
	5	23	50	2.660		4	22	2	2.847		34	60	12	1.124
	6	24	38	2.612		5	22	48	2.796		0	18	31	3.150
	7	25	28	2.564		6	23	36	2.746		1	19	12	3.096
	8	26	19	2.517		7	24	26	2.695		2	19	54	3.044
	9	27	13	2.470		8	25	16	2.643		3	20	38	2.992
	10	28	7	2.422		9	26	8	2.594		4	21	22	2.940
	11	29	3	2.375		10	27	1	2.545		5	22	7	2.887
	12	30	0	2.326		11	27	57	2.495		6	22	56	2.836
	13	30	57	2.279		12	28	53	2.443		7	23	47	2.784
	14	31	58	2.231		13	29	52	2.392		8	24	37	2.731
	15	33	0	2.182		14	30	50	2.341		9	25	29	2.679
	16	34	2	2.132		15	31	50	2.290		10	26	22	2.627
	17	35	6	2.083		16	32	55	2.239		11	27	18	2.575
	18	36	13	2.031		17	34	0	2.186		12	28	16	2.522
	19	37	22	1.980		18	35	6	2.133		13	29	14	2.469
	20	38	33	1.929		19	36	14	2.080		14	30	12	2.416
	21	39	46	1.875		20	37	23	2.027		15	31	12	2.363
	22	41	1	1.822		21	38	35	1.971		16	32	15	2.310
	23	42	18	1.768		22	39	50	1.916		17	33	19	2.255
	24	43	40	1.713		23	41	5	1.866		18	34	24	2.200
	25	45	4	1.656		24	42	21	1.803		19	35	30	2.145
	26	46	30	1.599		25	43	42	1.747		20	36	39	2.088
	27	48	2	1.541		26	45	8	1.689		21	37	50	2.032
	28	49	39	1.481		27	46	34	1.629		22	39	2	1.975
	29	51	24	1.415		28	48	7	1.568		23	40	18	1.918
	30	53	16	1.346		29	49	45	1.504		24	41	35	1.860
	31	55	22	1.270		30	51	28	1.436		25	42	54	1.802
	32	57	56	1.177		31	53	20	1.365		26	44	16	1.742
3.00	0	19	28	3.000	3.10	32	56	10	1.284	3.20	27	45	41	1.680
	1	20	10	2.949		33	58	0	1.195		28	47	13	1.618
	2	20	53	2.900		34	61	32	1.073		29	48	46	1.555
	3	21	37	2.849		0	18	49	3.100		30	50	27	1.489
	4	22	22	2.800		1	19	29	3.046		31	52	15	1.419
	5	23	8	2.750		2	20	13	2.995		32	54	13	1.343
	6	23	56	2.700		3	20	58	2.944		33	56	28	1.260
	7	24	45	2.651		4	21	43	2.892		34	59	11	1.160
	8	25	35	2.602		5	22	28	2.840		35	64	36	.990
	9	26	27	2.553		6	23	16	2.787		0	18	13	3.200
	10	27	21	2.503		7	24	7	2.736		1	18	54	3.146
	11	28	17	2.454		8	24	58	2.686		2	19	36	3.092
	12	29	13	2.404		9	25	48	2.635		3	20	20	3.040
	13	30	12	2.355		10	26	41	2.585		4	21	4	2.987
	14	31	12	2.304		11	27	37	2.533		5	21	50	2.933
	15	32	13	2.254		12	28	34	2.483		6	22	38	2.879
	16	33	14	2.203		13	29	32	2.431		7	23	28	2.826
	17	34	19	2.151		14	30	31	2.379		8	24	19	2.773
	18	35	25	2.100		15	31	31	2.327		9	25	10	2.720
	19	36	32	2.046		16	32	33	2.274		10	26	2	2.667
	20	37	43	1.993		17	33	36	2.220		11	26	58	2.614
	21	38	56	1.939		18	34	43	2.166		12	27	56	2.561
	22	40	9	1.884		19	35	51	2.112		13	28	54	2.507
	23	41	25	1.828		20	37	1	2.058		14	29	52	2.452
	24	42	44	1.773		21	38	12	2.002					

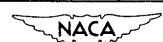


TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
 AND MACH NUMBER BEHIND THE SHOCK - Continued

M ₁	δ (deg)	ϵ		M ₂	M ₁	δ (deg)	ϵ		M ₂	M ₁	δ (deg)	ϵ		M ₂
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
3.20	15	30	52	2.398	3.30	7	22	52	2.910	3.35	36	63	24	1.033
	16	31	54	2.344		8	23	42	2.856		0	17	6	3.400
	17	33	0	2.288		9	24	33	2.801		1	17	49	3.341
	18	34	5	2.232		10	25	26	2.748		2	18	32	3.284
	19	35	10	2.176		11	26	23	2.692		3	19	14	3.226
	20	36	18	2.120		12	37	19	2.636		4	19	58	3.169
	21	37	32	2.062		13	28	16	2.580		5	20	44	3.111
	22	38	44	2.005		14	29	16	2.524		6	21	32	3.054
	23	39	58	1.946		15	30	18	2.467		7	22	21	2.996
	24	41	16	1.887		16	31	20	2.410		8	23	10	2.937
	25	42	34	1.829		17	32	22	2.355		9	24	2	2.881
	26	43	57	1.770		18	33	28	2.297		10	24	55	2.824
	27	45	22	1.709		19	34	35	2.239		11	25	50	2.766
	28	46	50	1.646		20	35	43	2.182		12	26	45	2.708
	29	48	22	1.583		21	36	53	2.123		13	27	44	2.651
	30	50	0	1.516		22	38	6	2.064		14	28	44	2.595
	31	51	45	1.445		23	39	21	2.005		15	29	44	2.536
	32	53	40	1.370		24	40	36	1.945		16	30	46	2.477
	33	55	50	1.289		25	41	54	1.884		17	31	50	2.420
	34	58	24	1.196		26	43	13	1.823		18	32	54	2.360
	35	62	6	1.067		27	44	34	1.760		19	34	2	2.301
3.25	0	17	55	3.250	3.35	29	47	34	1.631	3.45	20	35	10	2.241
	1	18	36	3.195		30	49	10	1.565		21	36	19	2.180
	2	19	18	3.140		31	50	50	1.496		22	37	30	2.120
	3	20	1	3.086		32	52	40	1.423		23	38	44	2.058
	4	20	46	3.031		33	54	42	1.344		24	40	0	1.997
	5	21	30	2.977		34	57	1	1.256		25	41	16	1.935
	6	22	20	2.923		35	59	54	1.152		26	42	35	1.873
	7	23	10	2.868		24	1	2.815	3.350		27	43	58	1.809
	8	24	52	2.760		0	17	22	3.294		28	45	24	1.744
	9	24	44	2.706		1	18	3	3.237		29	46	54	1.677
	10	25	40	2.651		2	18	45	3.181		30	48	25	1.610
	11	26	36	2.597		3	19	28	3.125		31	50	4	1.540
	12	27	34	2.543		4	20	0	3.068		32	51	50	1.468
	13	28	34	2.489		5	21	49	3.013		33	53	42	1.392
	14	29	35	2.433		6	21	37	2.955		34	55	53	1.310
	15	30	38	2.378		7	22	26	2.899		35	58	24	1.214
	16	31	41	2.322		8	23	18	2.842		36	61	54	1.090
	17	32	46	2.265		9	24	10	2.786		0	16	51	3.450
	18	33	53	2.210		10	25	6	2.729		1	17	32	3.391
	19	34	1	2.152		11	26	4	2.672		2	18	13	3.333
	20	36	13	2.094		12	27	2	2.615		3	18	56	3.275
	21	37	12	2.035		13	28	0	2.558		4	19	41	3.216
	22	38	24	1.975		14	29	30	2.500		5	20	26	3.156
	23	39	38	1.916		15	30	3	2.442		6	21	14	3.097
	24	40	54	1.866		16	31	6	2.385		7	22	4	3.040
	25	42	10	1.806		17	32	12	2.327		8	22	54	2.981
	26	43	31	1.759		18	33	19	2.269		9	23	46	2.922
	27	44	55	1.734		19	34	25	2.211		10	24	39	2.864
	28	46	25	1.671		20	35	34	2.152		11	25	34	2.805
	29	47	57	1.607		21	36	46	2.092		12	26	30	2.747
	30	49	33	1.540		22	37	2	2.032		13	27	27	2.687
	31	51	18	1.469		23	39	18	1.970		14	28	26	2.630
	32	53	8	1.398		24	40	35	1.910		15	29	28	2.570
	33	55	14	1.318		25	41	54	1.846		16	30	30	2.512
	34	57	38	1.228		26	42	18	1.783		17	31	34	2.452
	35	60	50	1.116		27	44	44	1.719		18	32	37	2.391
3.30	0	17	35	3.300	3.45	28	45	14	1.655	3.45	19	33	45	2.331
	1	18	18	3.244		29	47	46	1.587		20	34	51	2.270
	2	19	2	3.189		30	48	28	1.520		21	36	0	2.208
	3	19	46	3.132		31	50	14	1.445		22	37	14	2.147
	4	20	30	3.077		32	52	10	1.369		23	38	27	2.085
	5	21	15	3.022		33	54	24	1.283		24	39	43	2.022
	6	22	4	2.966		35	59	4	1.186		25	41	0	1.958



TABLE III - Continued

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,

AND MACH NUMBER BEHIND THE SHOCK - Continued

M ₁	δ (deg)	ε		M ₂	M ₁	δ (deg)	ε		M ₂	M ₁	δ (deg)	ε		M ₂
		(deg)	(min)				(deg)	(min)				(deg)	(min)	
3.45	26	42	18	1.895	3.55	16	29	59	2.576	3.65	4	18	41	3.396
	27	43	38	1.831		17	31	3	2.516		5	19	28	3.333
	28	45	5	1.766		18	32	6	2.455		6	20	15	3.270
	29	46	34	1.700		19	33	12	2.392		7	21	4	3.207
	30	48	6	1.631		20	34	20	2.330		8	21	55	3.145
	31	49	42	1.562		21	35	30	2.266		9	22	46	3.082
	32	51	26	1.491		22	36	42	2.203		10	23	40	3.020
	33	53	17	1.416		23	37	55	2.139		11	24	34	2.957
	34	55	22	1.334		24	39	10	2.075		12	25	30	2.894
	35	57	45	1.242		25	40	26	2.009		13	26	28	2.831
	36	61	0	1.127		26	41	44	1.943		14	27	28	2.766
						27	43	8	1.877		15	28	28	2.703
						28	44	32	1.812		16	29	30	2.640
						29	45	57	1.745		17	30	34	2.577
3.50	0	16	36	3.500		30	47	28	1.675		18	31	38	2.513
	1	17	16	3.440		31	49	2	1.606		19	32	46	2.448
	2	17	58	3.381		32	50	44	1.535		20	33	53	2.383
	3	18	40	3.321		33	52	30	1.459		21	35	2	2.317
	4	19	24	3.262		34	54	29	1.380		22	36	14	2.252
	5	20	10	3.203		35	56	42	1.291		23	37	26	2.187
	6	21	0	3.141		36	59	26	1.187		24	38	40	2.121
	7	21	48	3.082		37	64	10	1.019		25	39	56	2.055
	8	22	38	3.023		38	64				26	41	14	1.987
	9	23	29	2.963		39	64				27	42	34	1.922
	10	24	24	2.904	3.60	0	16	8	3.600		28	43	57	1.855
	11	25	18	2.844		1	16	47	3.538		29	45	24	1.786
	12	26	14	2.785		2	17	29	3.474		30	46	54	1.718
	13	27	10	2.724		3	18	12	3.412		31	48	26	1.648
	14	28	10	2.662		4	18	56	3.350		32	50	4	1.575
	15	29	11	2.602		5	19	41	3.288		33	51	49	1.500
	16	30	14	2.543		6	20	30	3.226		34	53	44	1.422
	17	31	17	2.483		7	21	19	3.166		35	55	46	1.337
	18	32	22	2.422		8	22	10	3.104		36	58	16	1.241
	19	33	28	2.361		9	23	0	3.044		37	60	50	1.117
	20	34	35	2.299		10	23	54	2.981		3.70	0	15	41
	21	35	45	2.237		11	24	48	2.919		1	16	20	3.635
	22	36	57	2.174		12	25	44	2.857		2	17	2	3.569
	23	38	10	2.111		13	26	42	2.795		3	17	44	3.503
	24	39	24	2.048		14	27	41	2.731		4	18	28	3.438
	25	40	42	1.983		15	28	41	2.670		5	19	15	3.376
	26	42	2	1.918		16	29	45	2.608		6	20	2	3.312
	27	43	24	1.852		17	30	47	2.546		7	21	50	3.250
	28	44	47	1.787		18	31	52	2.483		8	22	41	3.186
	29	46	14	1.720		19	32	58	2.419		9	23	33	3.124
	30	47	46	1.652		20	34	6	2.357		10	23	26	3.060
	31	49	21	1.584		21	35	15	2.292		11	24	21	2.996
	32	51	4	1.513		22	36	27	2.227		12	25	17	2.932
	33	52	53	1.437		23	37	39	2.163		13	26	14	2.868
	34	54	55	1.356		24	38	53	2.097		14	27	14	2.802
	35	57	14	1.266		25	40	10	2.032		15	28	15	2.737
	36	60	12	1.158		26	41	28	1.966		16	29	16	2.672
3.55	0	16	22	3.550		27	42	49	1.900		17	30	20	2.606
	1	17	2	3.488		28	44	13	1.834		18	33	26	2.540
	2	17	44	3.426		29	45	39	1.766		19	34	49	2.475
	3	18	26	3.365		30	48	8	1.699		20	35	40	2.410
	4	19	10	3.304		31	48	43	1.628		21	36	49	2.342
	5	19	56	3.244		32	50	24	1.556		22	37	0	2.277
	6	20	46	3.184		33	52	9	1.481		23	38	28	2.211
	7	21	34	3.125		34	54	7	1.402		24	39	42	2.145
	8	22	25	3.065		35	56	14	1.315		25	40	42	2.077
	9	23	15	3.005		36	58	49	1.215		26	41	2	2.010
	10	24	10	2.943		37	62	39	1.075		27	42	22	1.943
	11	25	4	2.882	3.65	0	15	54	3.650		28	43	45	1.875
	12	26	0	2.821		1	16	34	3.587		29	44	10	1.806
	13	26	56	2.760		2	17	16	3.523		30	45	38	1.737
	14	27	54	2.699		3	17	58	3.460					
	15	28	56	2.637										



TABLE III - Concluded

VALUES OF MACH NUMBER, DEVIATION OF FLOW, ANGLE OF SHOCK,
 AND MACH NUMBER BEHIND THE SHOCK - Concluded

M ₁	δ (deg)	ε		M ₂	M ₁	δ (deg)	ε		M ₂
		(deg)	(min)				(deg)	(min)	
3.70	31	48	10	1.665	3.90	15	27	25	2.864
	32	49	46	1.594		16	28	26	2.796
	33	51	30	1.518		17	29	30	2.727
	34	53	21	1.440		18	30	34	2.657
	35	55	22	1.357		19	31	40	2.587
	36	57	46	1.263		20	32	48	2.517
	37	60	50	1.148		21	33	57	2.446
						22	35	8	2.377
						23	36	21	2.308
						24	37	36	2.237
						25	38	50	2.166
						26	40	8	2.095
						27	41	28	2.025
						28	42	50	1.955
3.80	0	15	15	3.800	4.0	29	44	13	1.885
	1	15	55	3.734		30	45	40	1.812
	2	16	36	3.665		31	47	10	1.741
	3	17	18	3.597		32	48	43	1.666
	4	18	2	3.532		33	50	22	1.591
	5	18	49	3.467		34	52	6	1.513
	6	19	36	3.402		35	54	2	1.431
	7	20	25	3.336		36	56	9	1.343
	8	21	16	3.270		37	58	38	1.240
	9	22	8	3.204		38	62	8	1.114
	10	23	1	3.137		0	14	29	4.000
	11	23	57	3.069		1	15	10	3.927
	12	24	53	3.002		2	15	50	3.854
	13	25	50	2.935		3	16	33	3.783
	14	26	19	2.866		4	17	17	3.711
	15	27	50	2.800		5	18	2	3.639
	16	28	50	2.733		6	18	50	3.568
	17	29	56	2.665		7	19	39	3.497
	18	31	2	2.598		8	20	30	3.426
	19	32	7	2.531		9	21	20	3.355
	20	33	15	2.462		10	22	13	3.285
	21	34	23	2.395		11	23	9	3.213
	22	35	34	2.328		12	24	6	3.141
	23	36	46	2.260		13	25	3	3.070
	24	38	0	2.193		14	26	2	2.999
	25	39	16	2.124		15	27	4	2.926
	26	40	34	2.055		16	28	6	2.856
	27	41	54	1.986		17	29	9	2.780
	28	43	16	1.916		18	30	14	2.713
	29	44	41	1.846		19	31	18	2.641
	30	46	8	1.776		20	32	26	2.570
	31	47	38	1.706		21	33	37	2.497
	32	49	13	1.631		22	34	46	2.424
	33	50	54	1.557		23	35	58	2.354
	34	52	41	1.480		24	37	12	2.282
	35	54	42	1.397		25	38	28	2.210
	36	56	57	1.306		26	39	46	2.137
	37	59	40	1.200		27	40	46	2.065
	38	64	14	1.026		28	41	5	1.992
3.90	0	14	51	3.900		23	35	58	1.922
	1	15	30	3.828		24	37	12	1.850
	2	16	12	3.757		25	38	28	1.777
	3	16	54	3.687		26	39	46	1.700
	4	17	37	3.618		27	41	5	1.625
	5	18	22	3.551		28	42	26	1.548
	6	19	10	3.482		29	43	48	1.465
	7	20	0	3.413		30	45	14	1.380
	8	20	51	3.345		31	46	42	1.281
	9	21	43	3.276		32	48	14	1.165
	10	22	36	3.207		33	49	55	
	11	23	32	3.139		34	51	38	
	12	24	29	3.070		35	53	27	
	13	25	26	3.002		36	55	30	
	14	26	24	2.932		37	57	52	
						38	60	50	

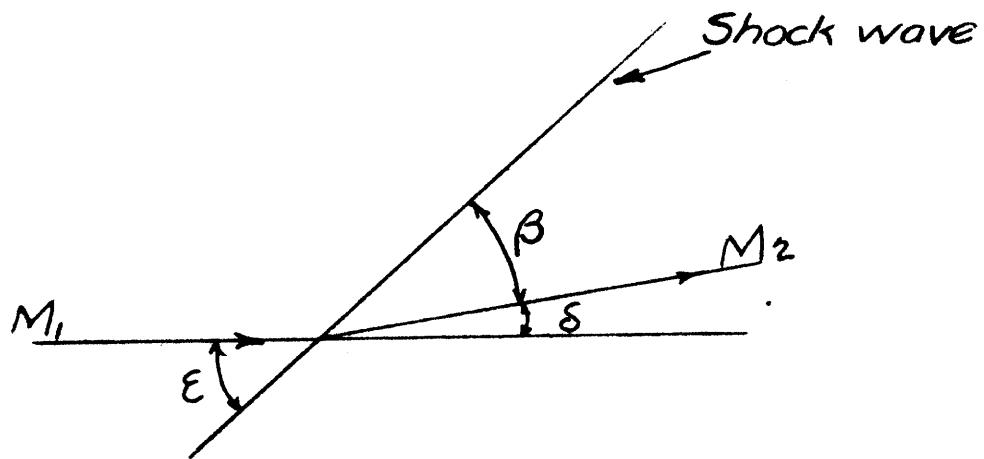


Figure 1.- Diagram showing the angle of the shock, deviation of the flow, and the angle of the flow behind the shock.

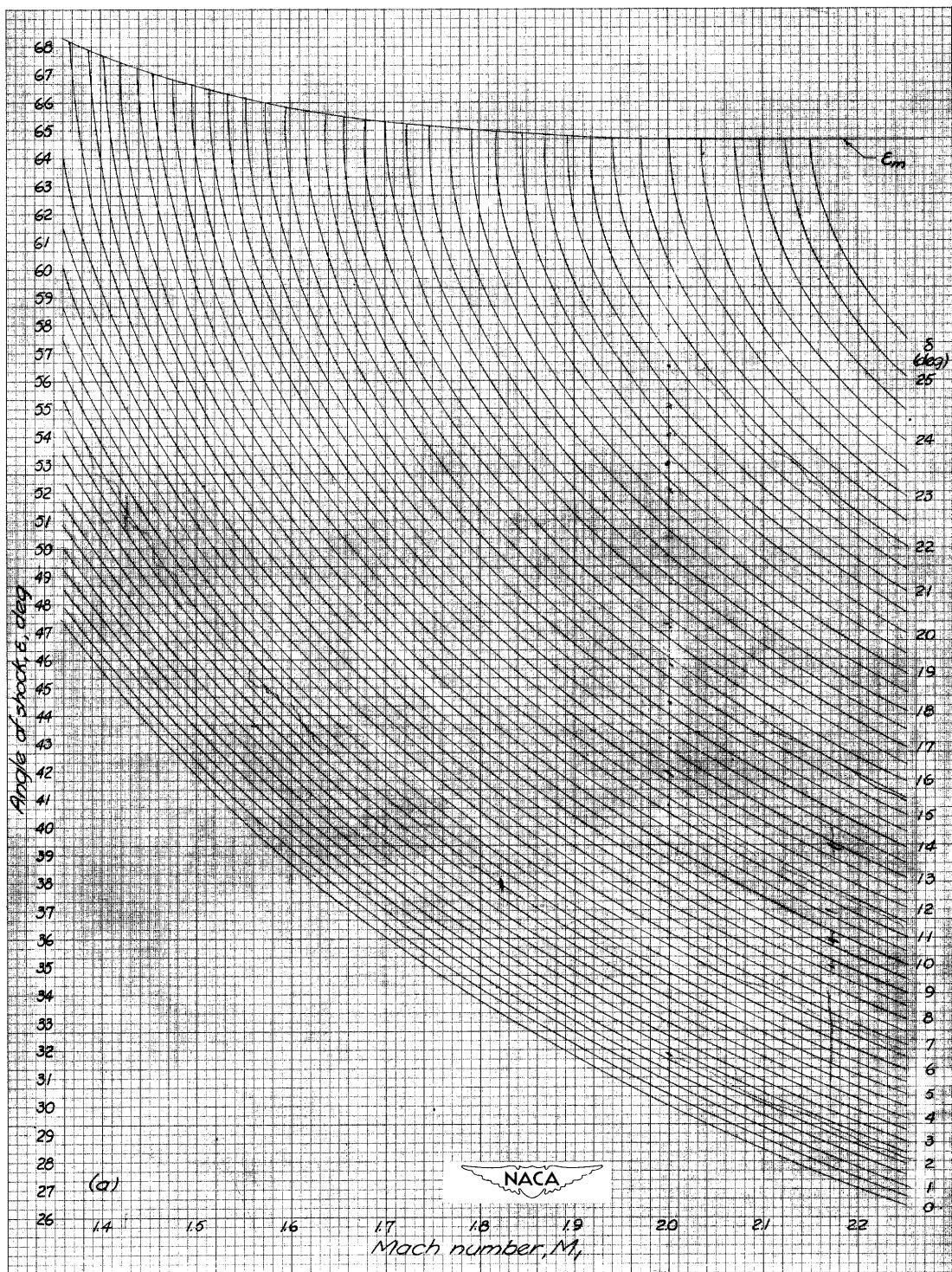


Figure 2.- Variation of angle of shock with Mach number in front of the shock, for various deviations of the flow.

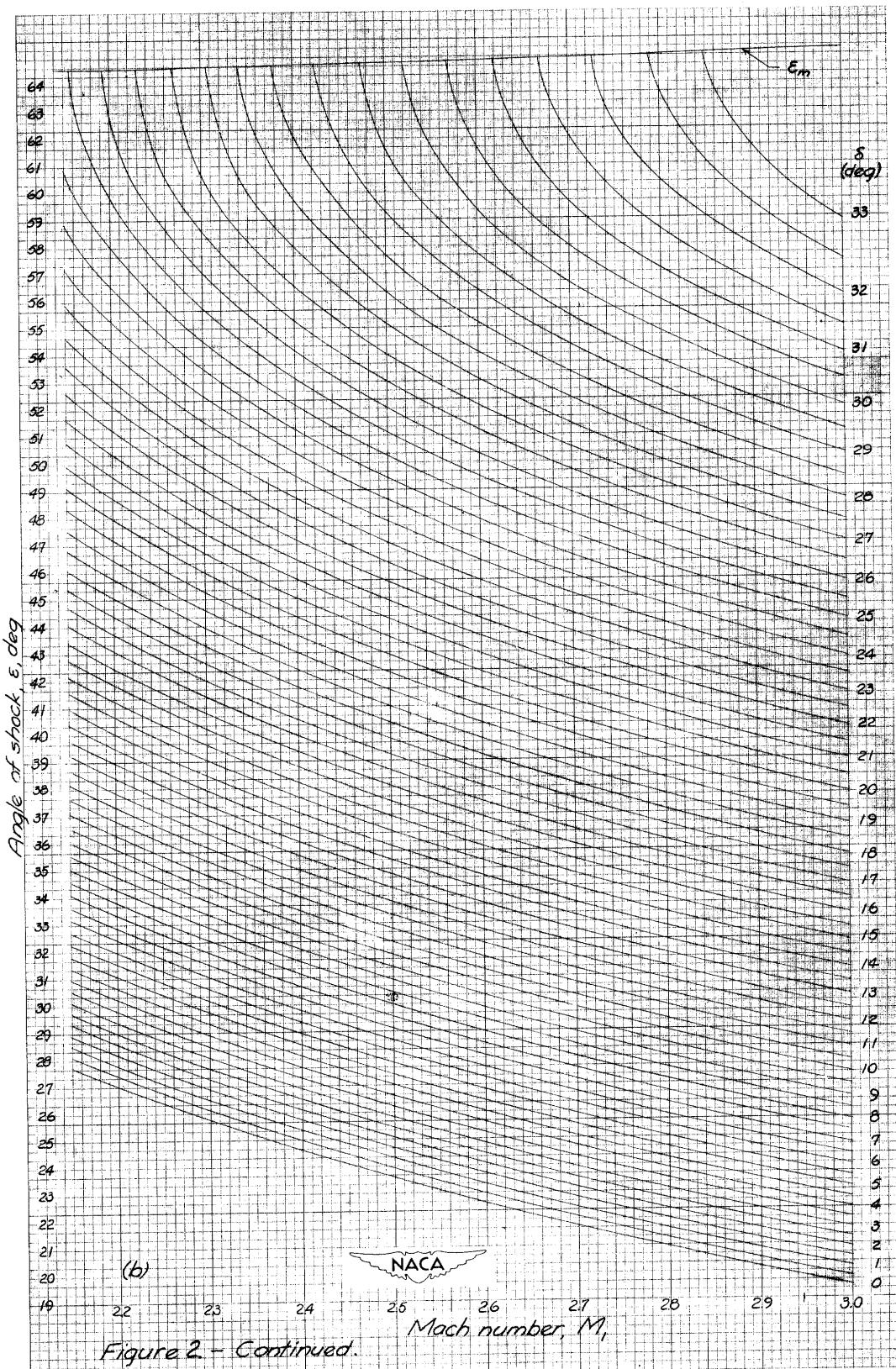


Figure 2 - Continued.

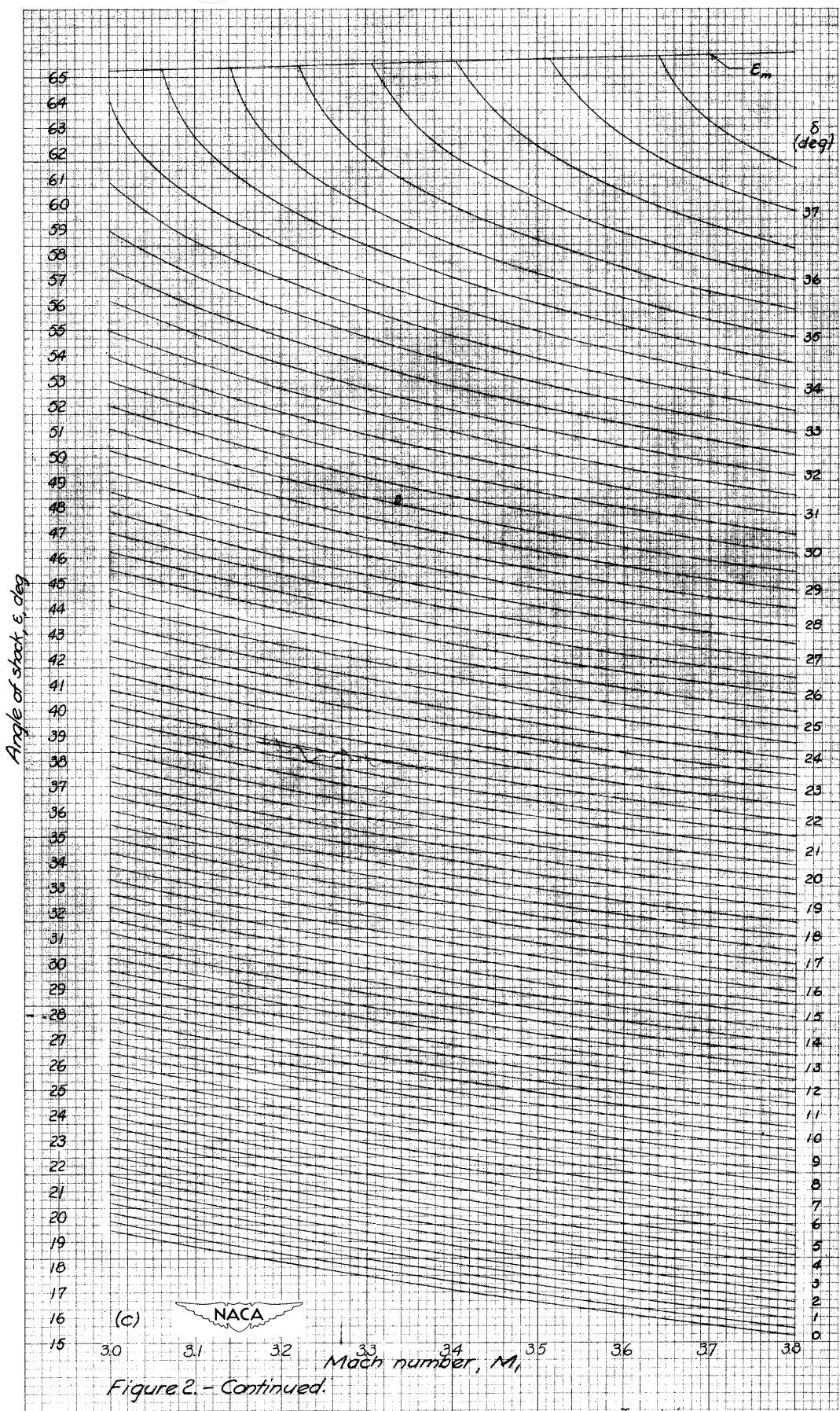
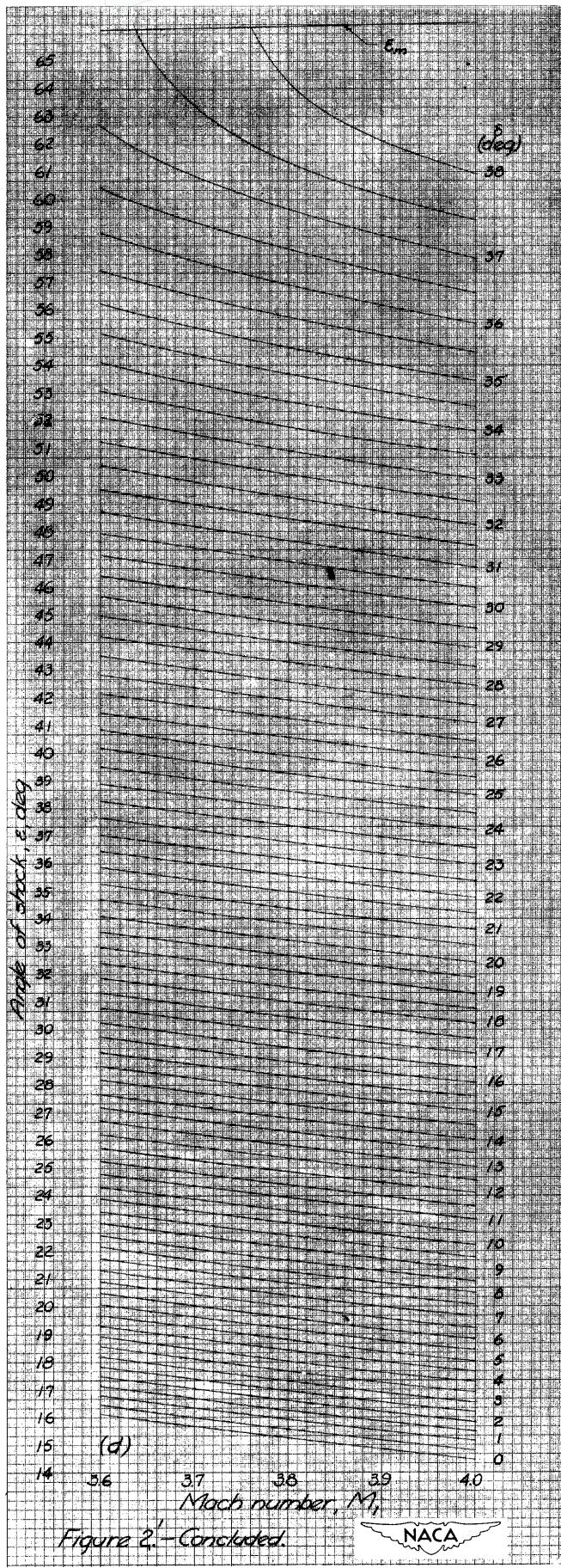


Figure 2. - Continued.



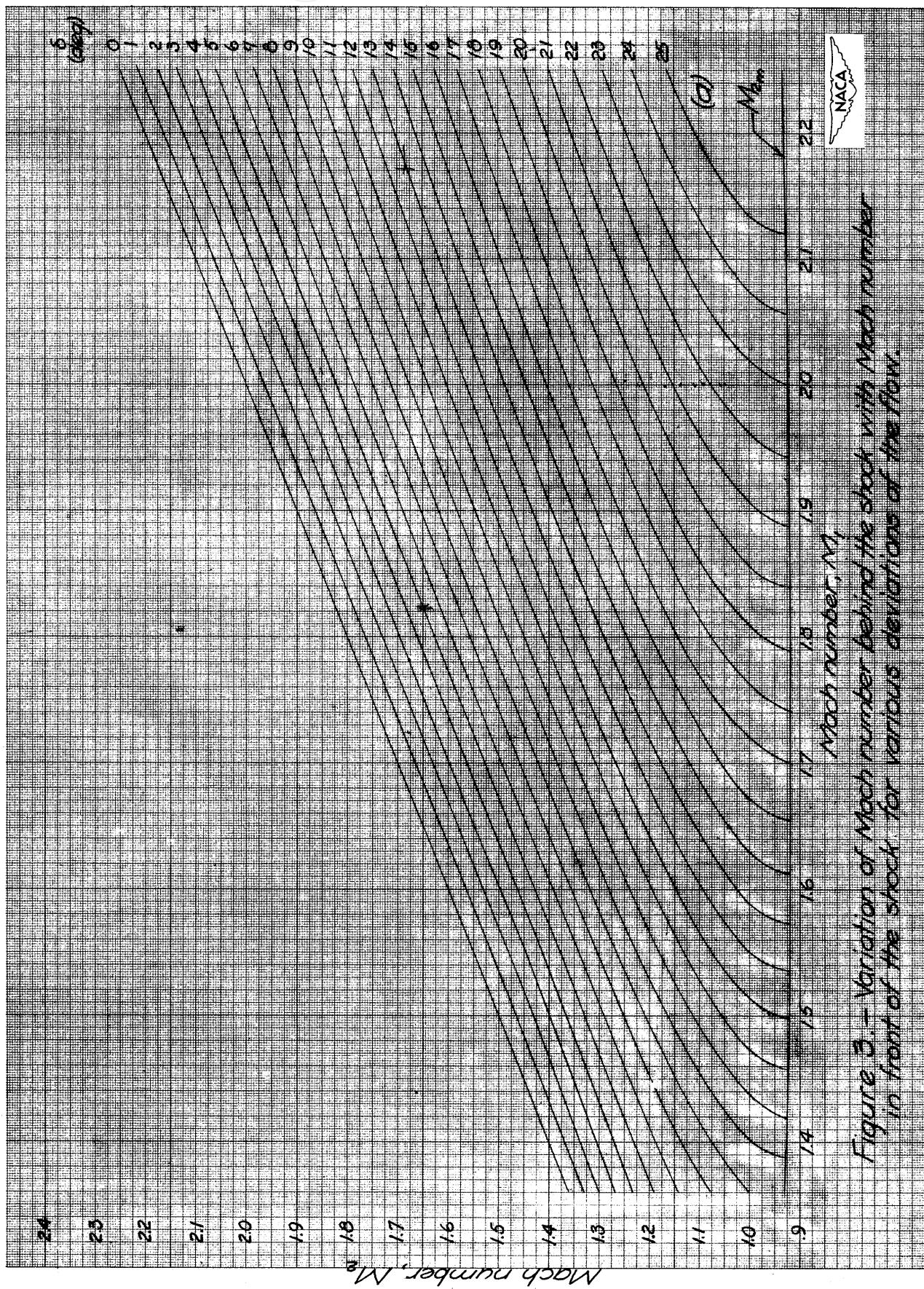


Figure 3.—Variation of moor number behind the stock wall, March number in front of the stock, for various definitions of the flow.

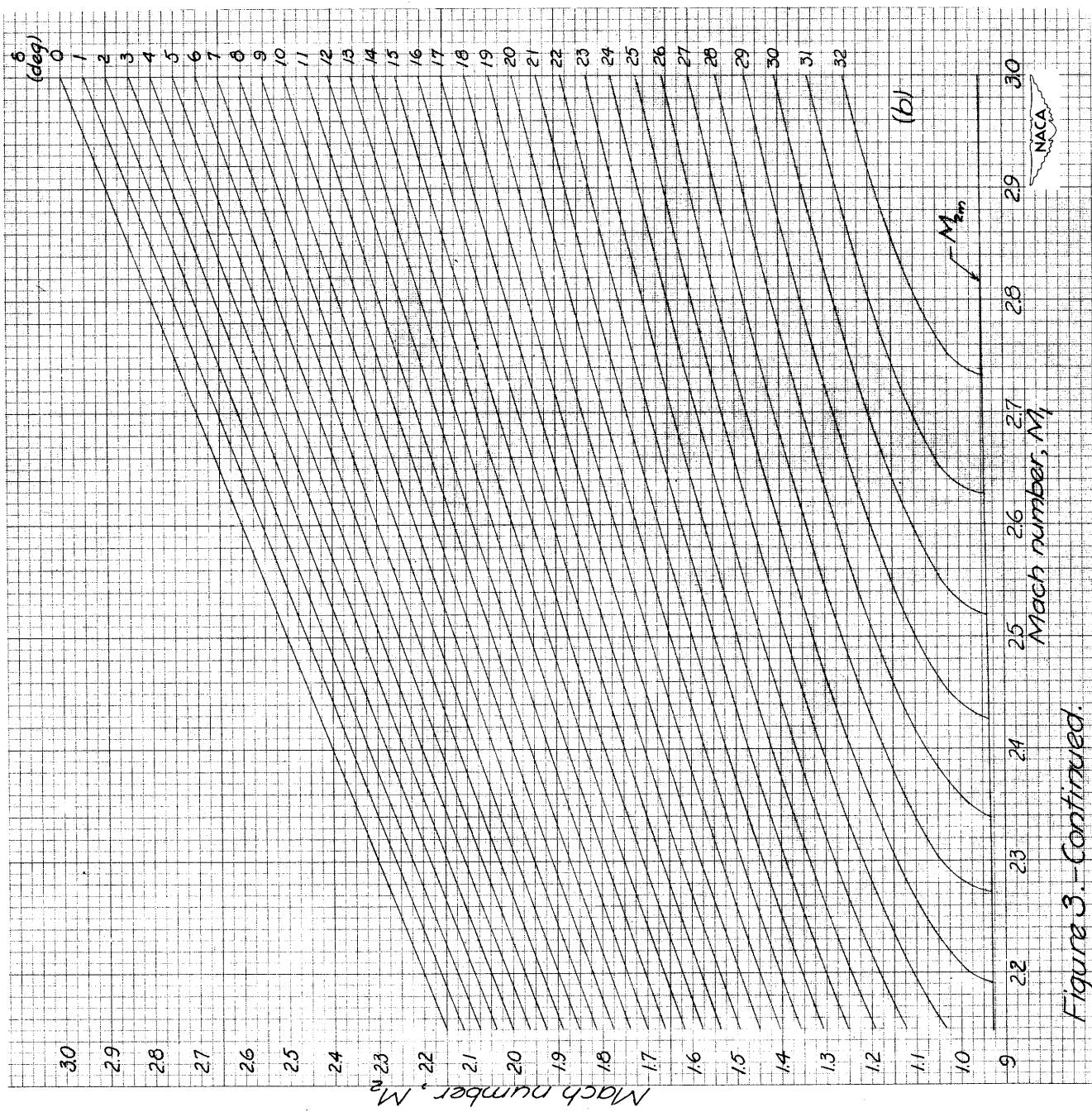


Figure 3—Continued.

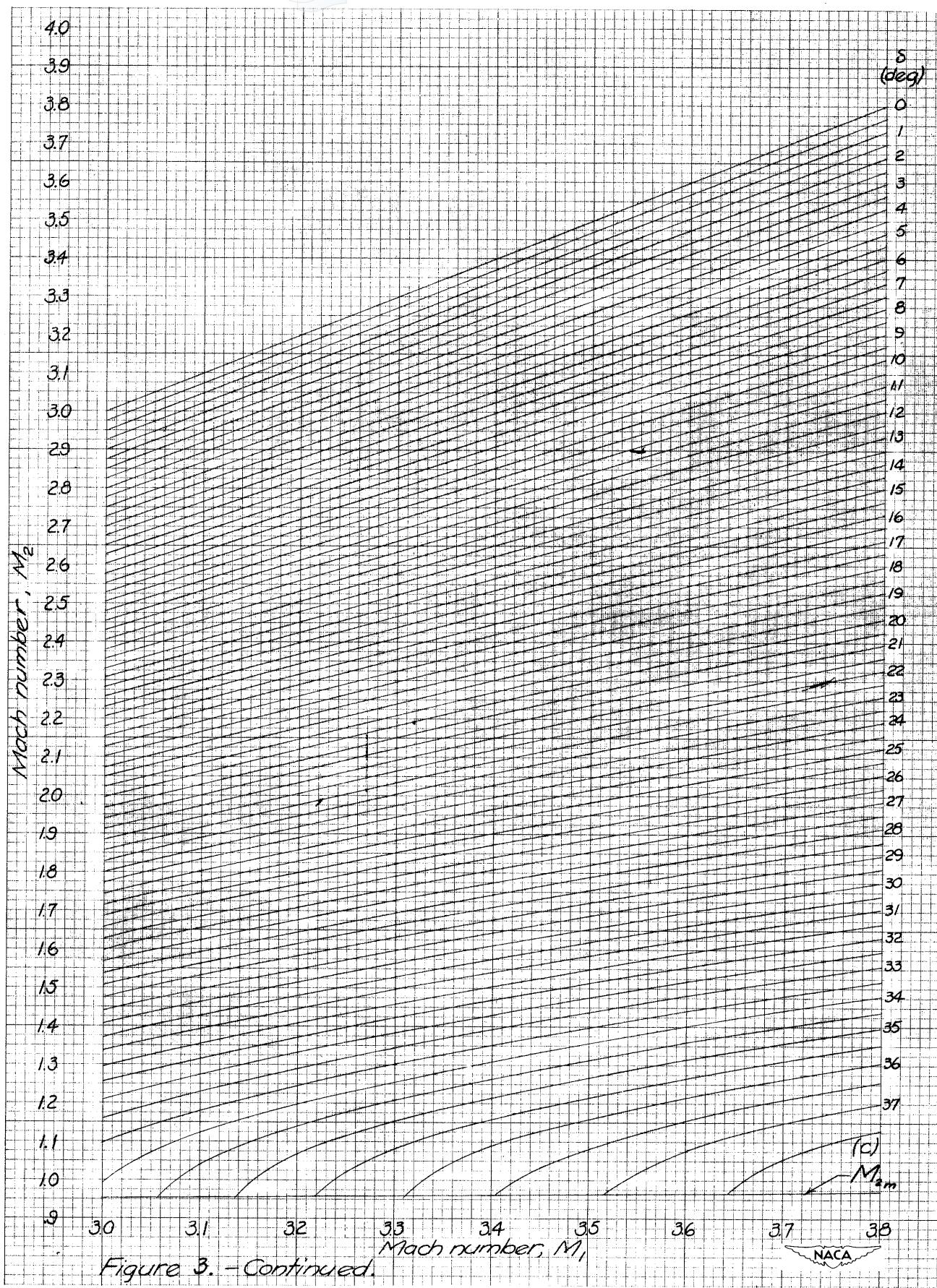
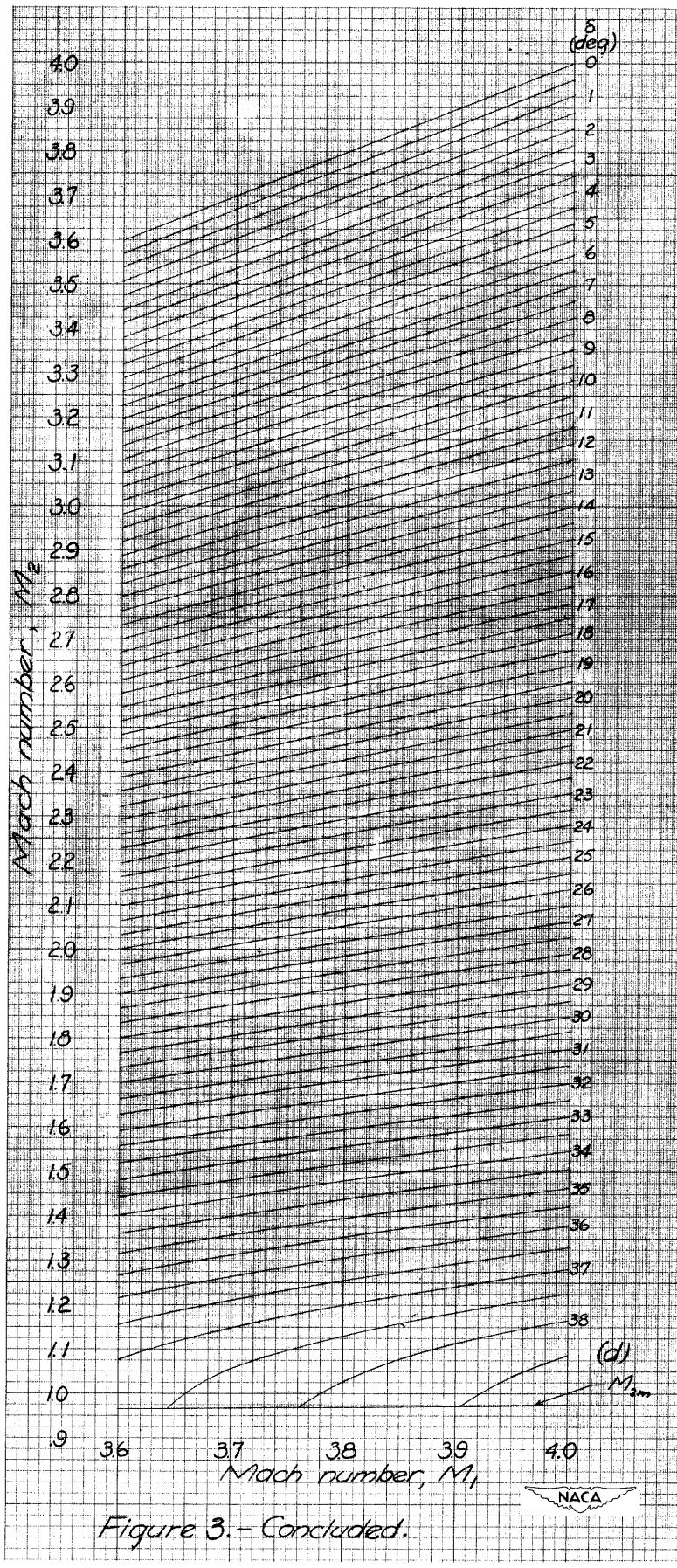


Figure 3. - Continued.



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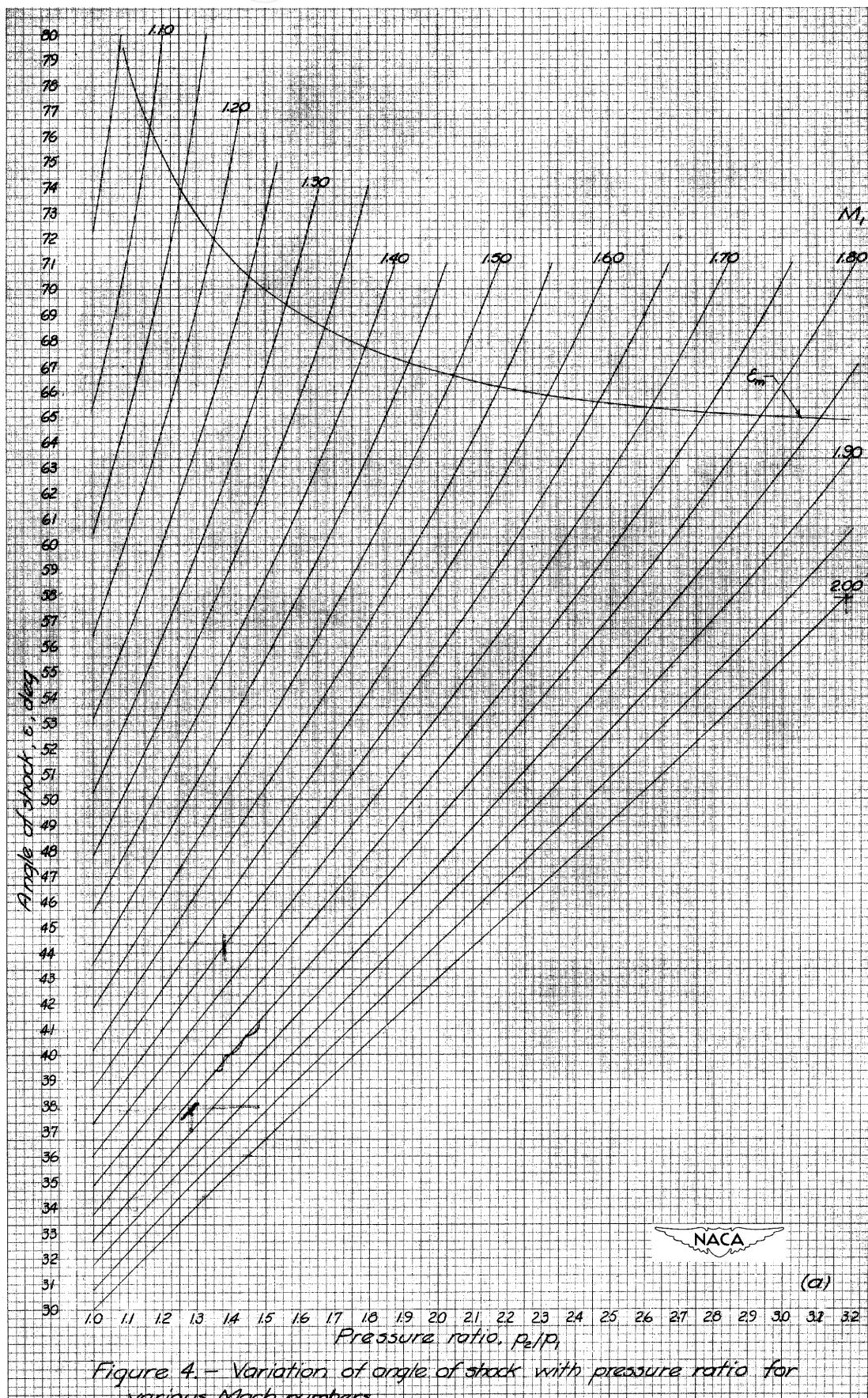


Figure 4.—Variation of angle of shock with pressure ratio for various Mach numbers.

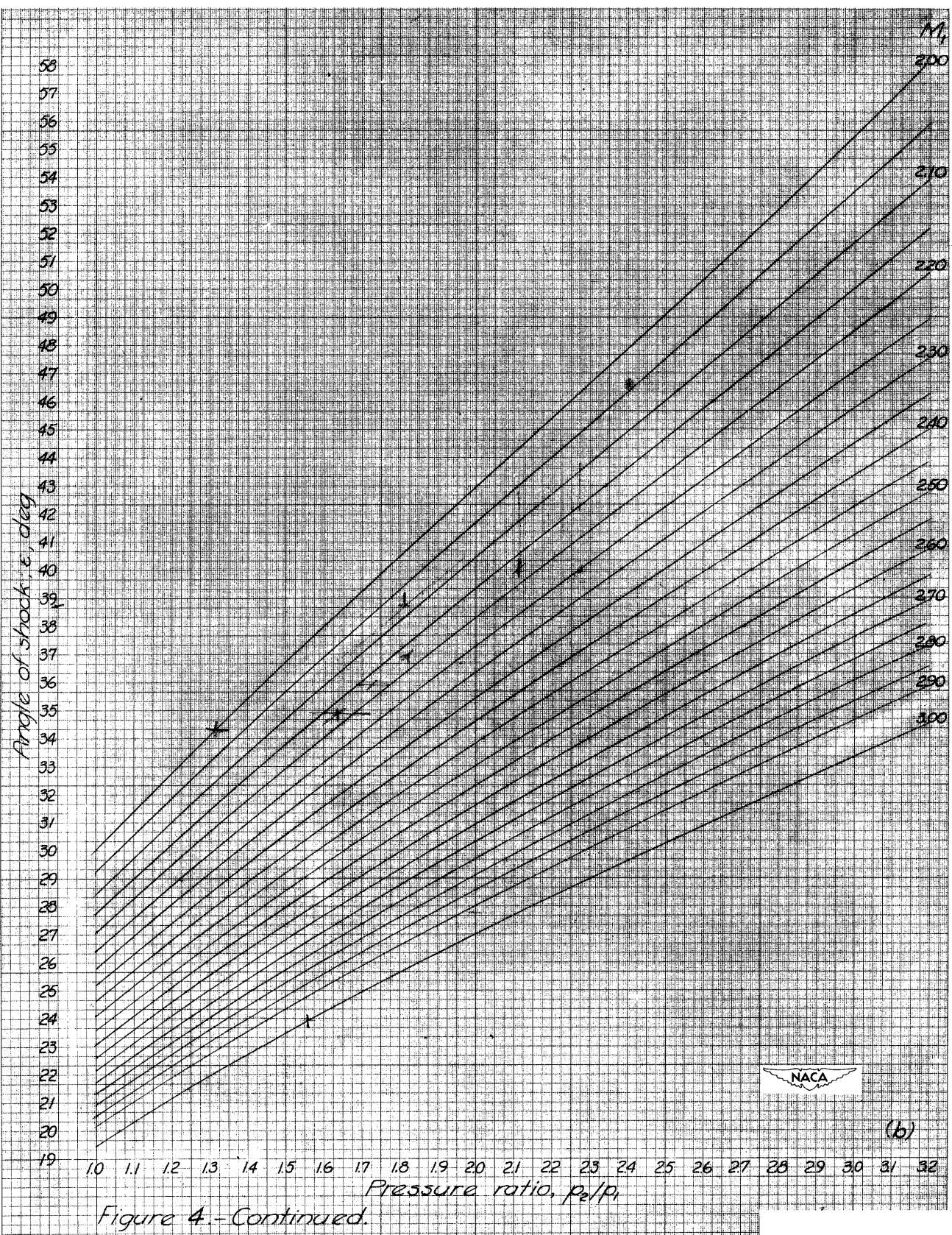
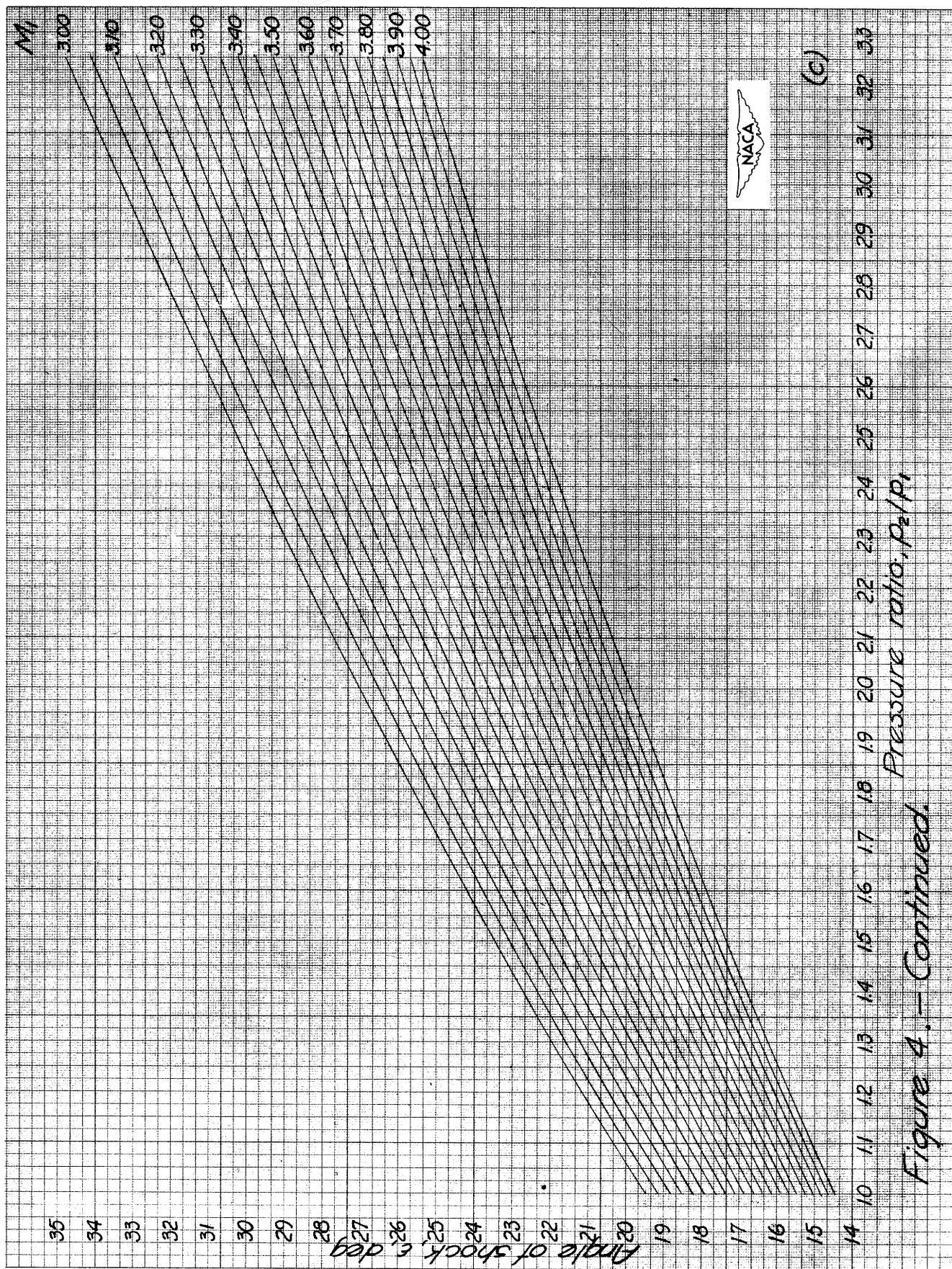


Figure 4.-Continued.



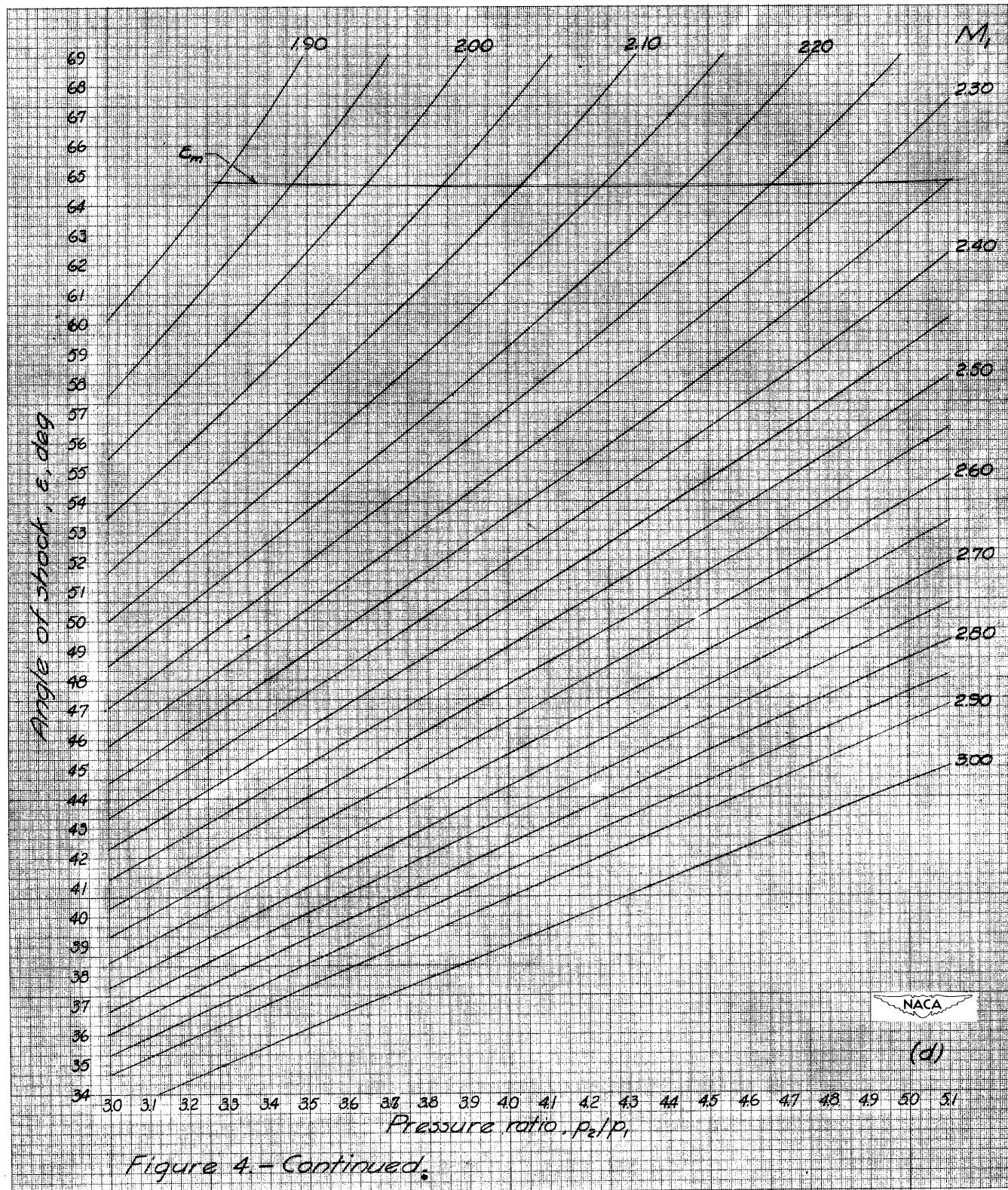


Figure 4 - Continued:

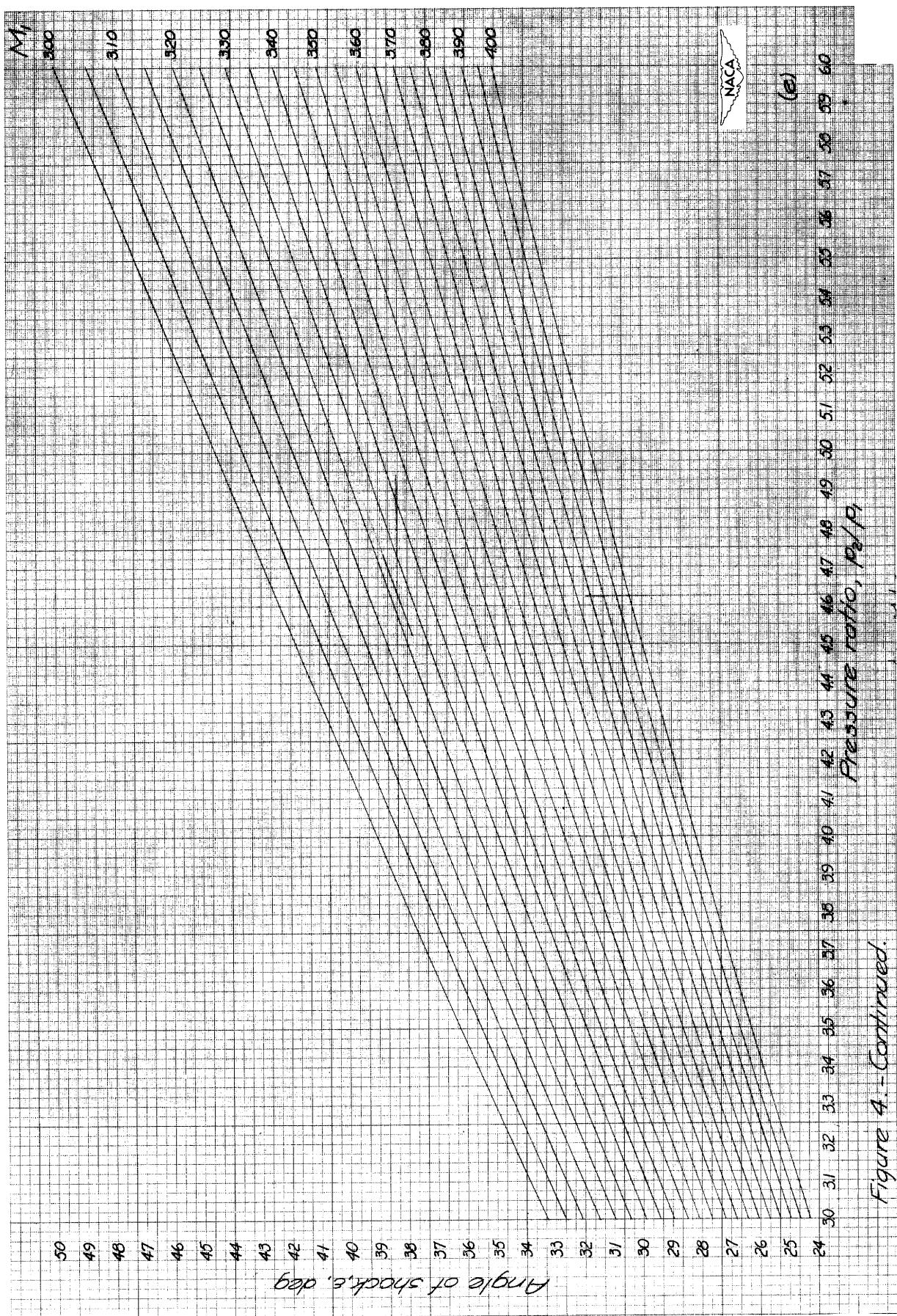


Figure 4 - Continued.

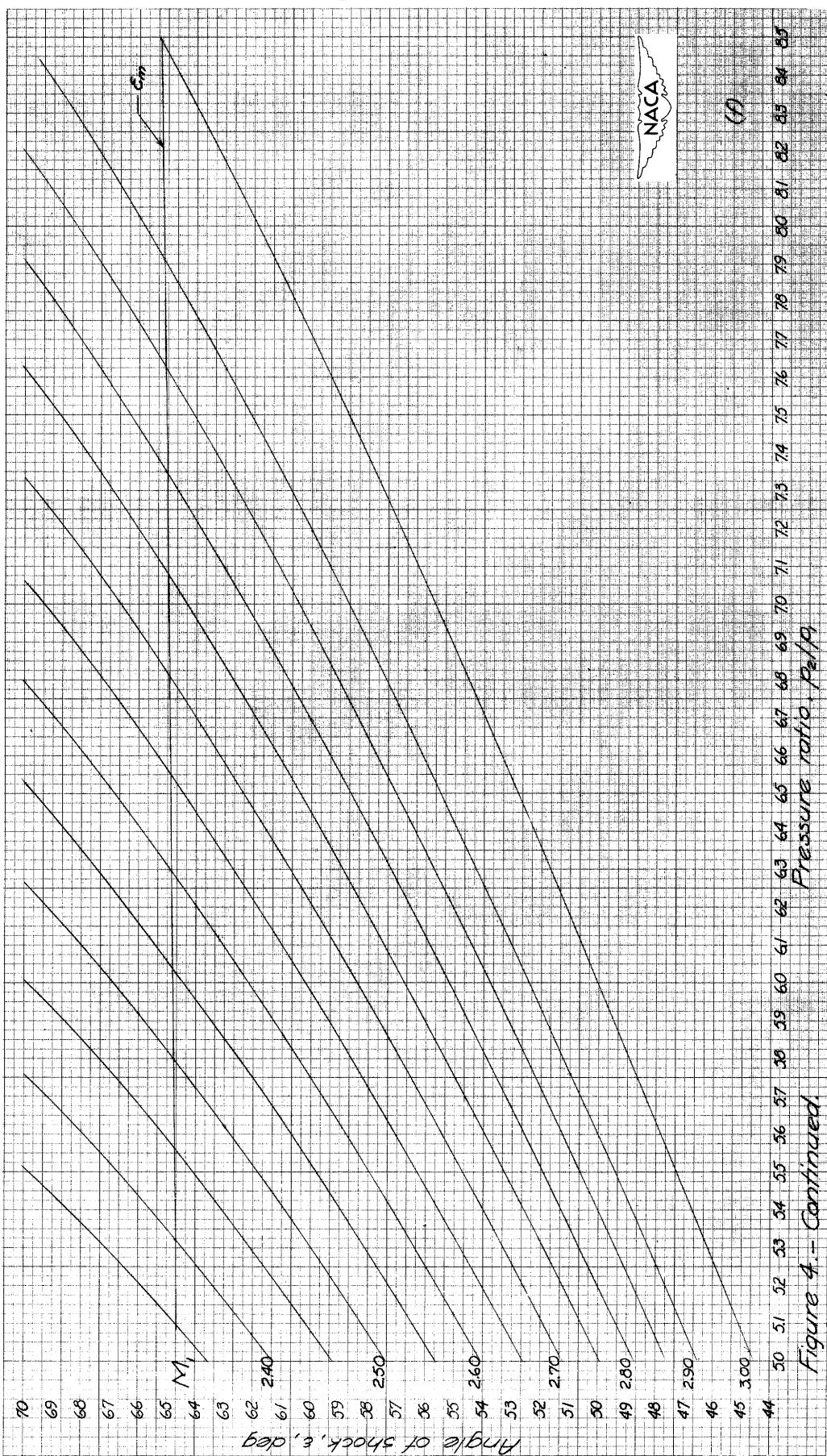
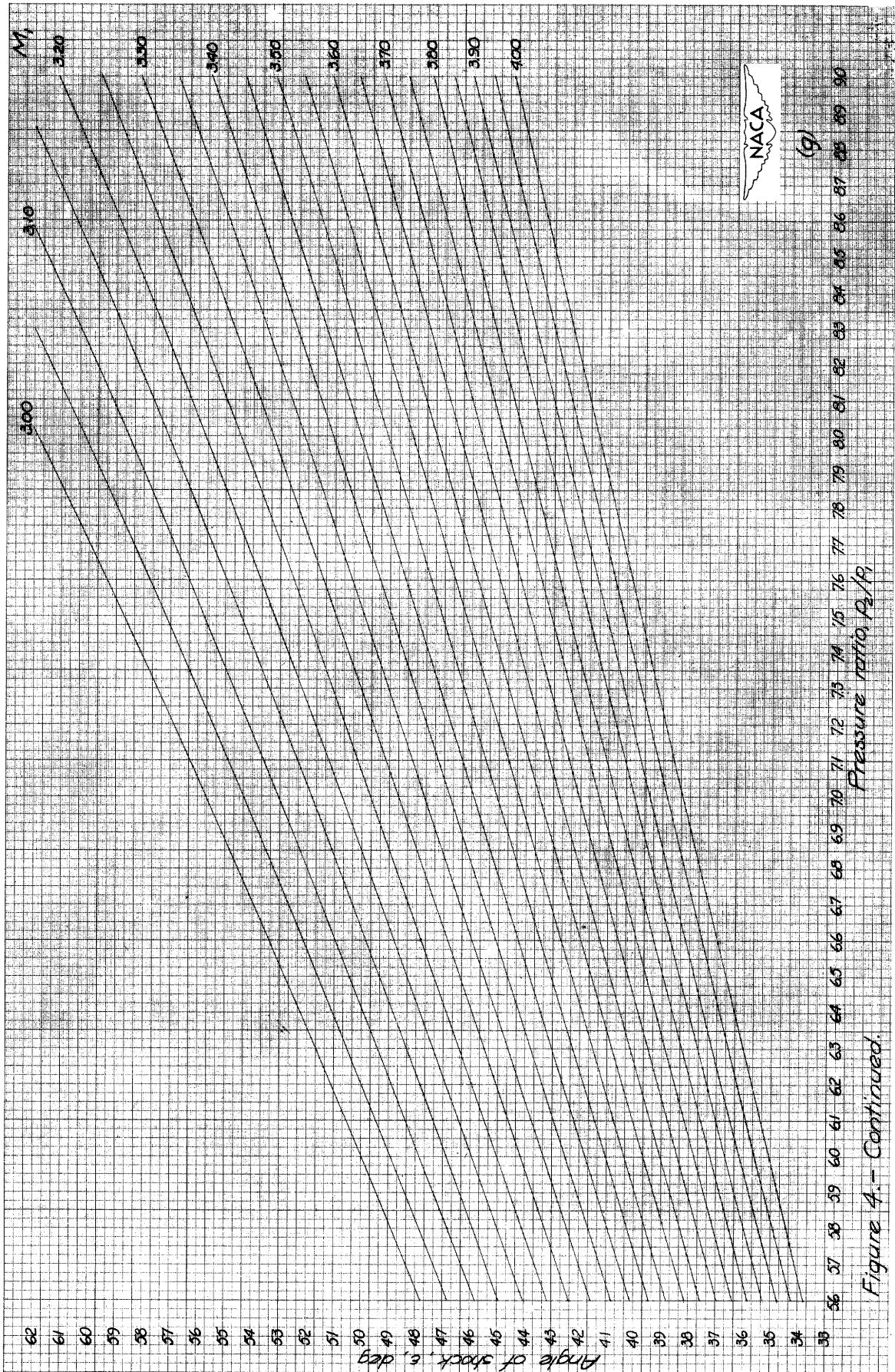


Figure 4 - Continued
Angle of shock, ϵ , deg
Pressure ratio, P_2/P_1



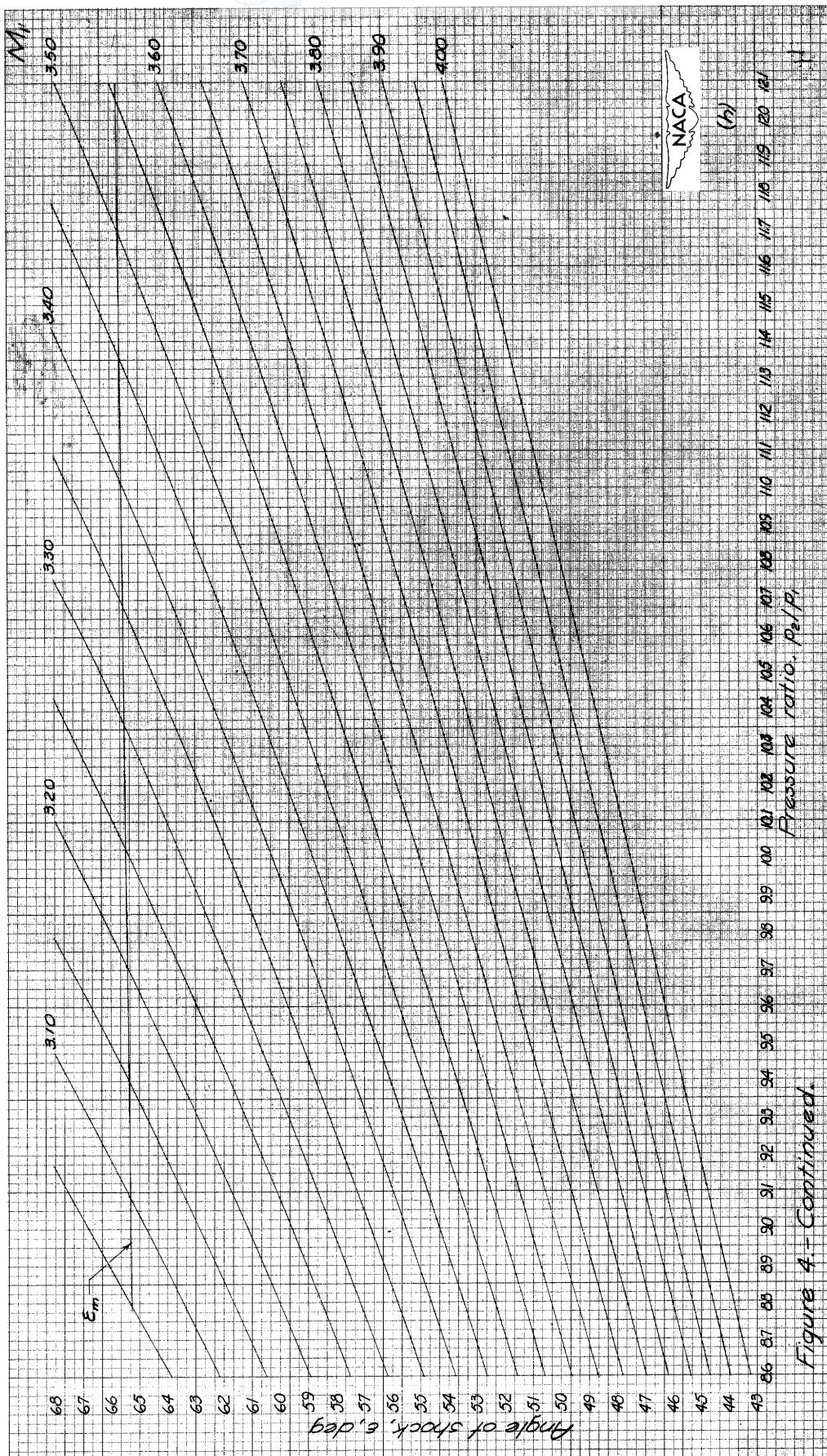
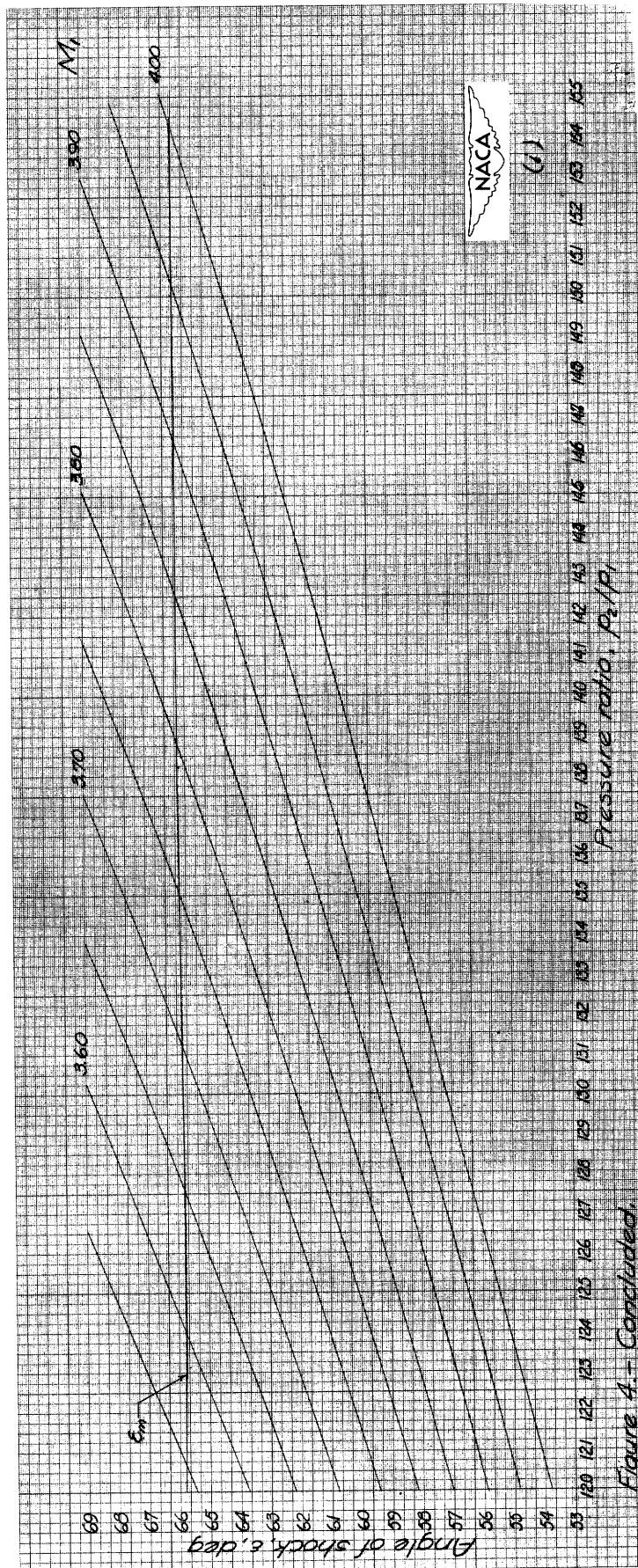


Figure 4-Continued.



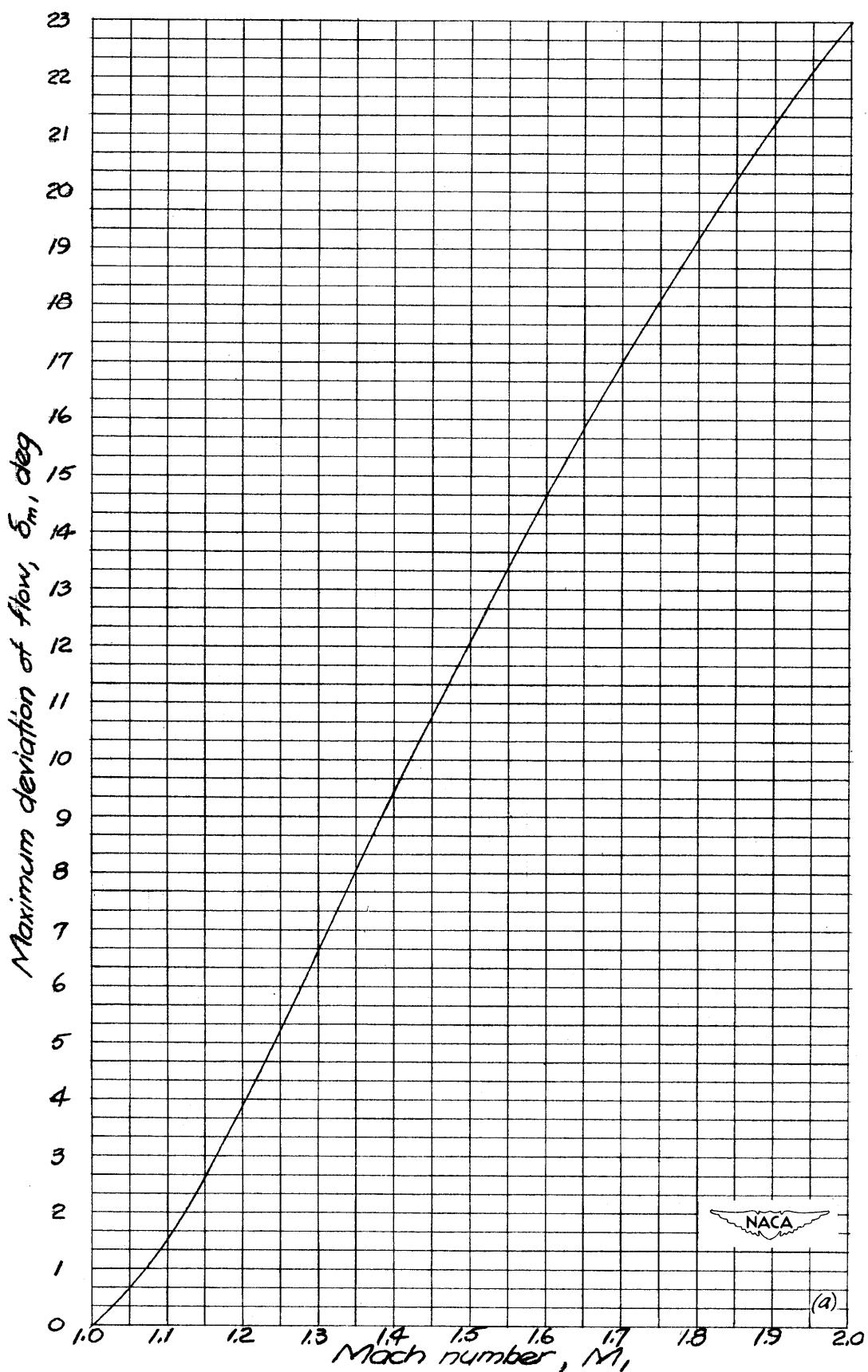


Figure 5.- Variation of maximum deviation of flow across the shock with Mach number in front of shock.

NACA TN No. 1673

43

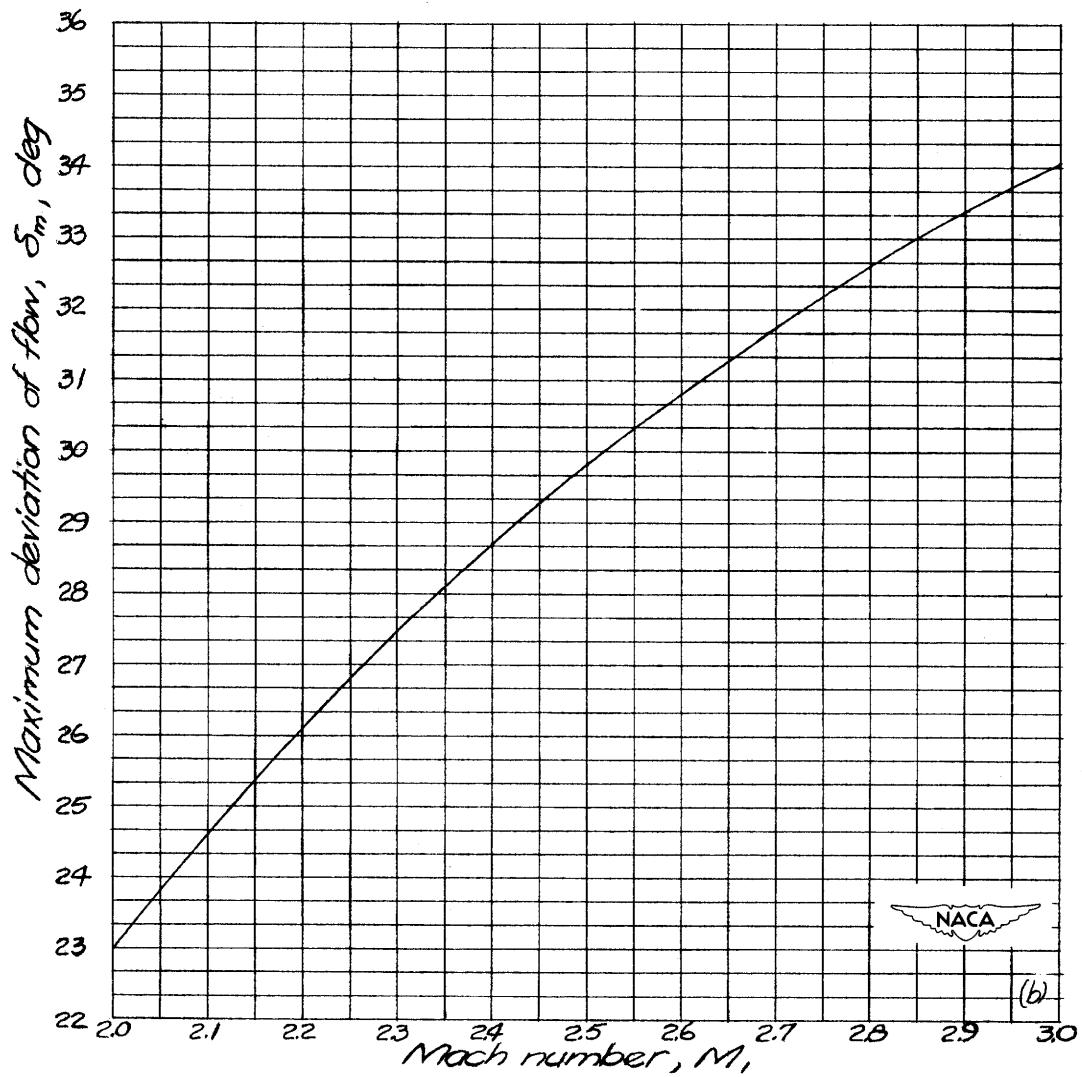


Figure 5.- Continued.

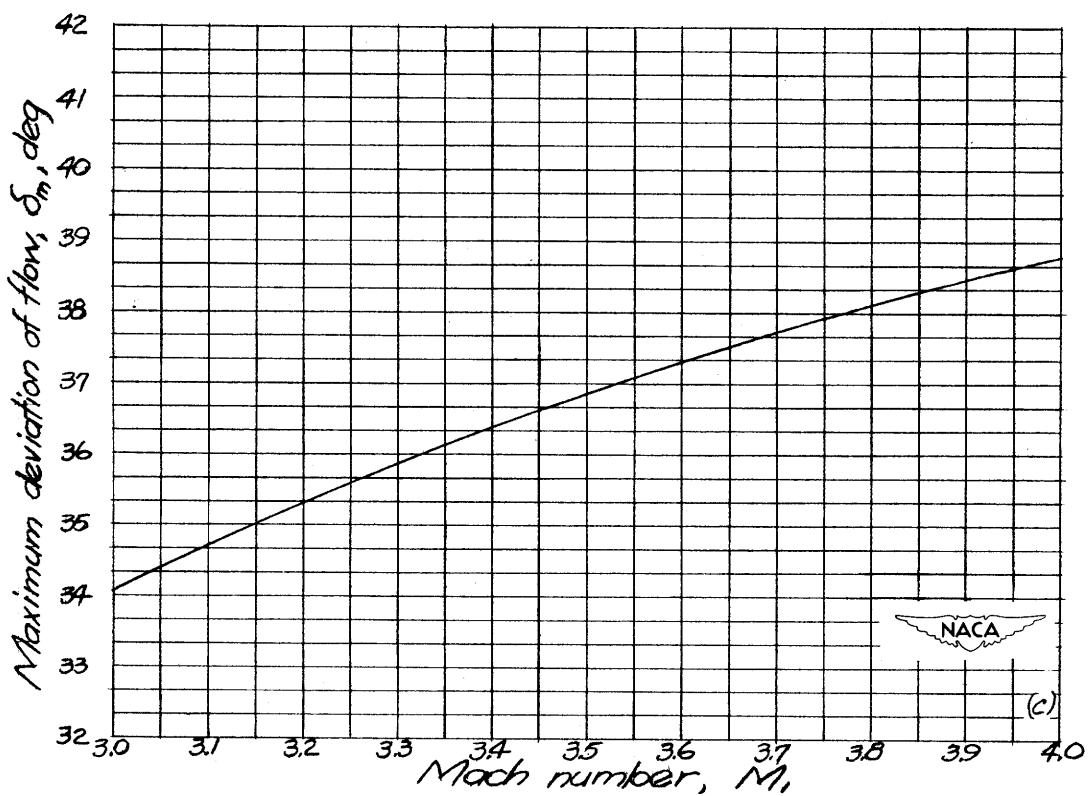


Figure 5.- Concluded.