

R. & M. No. 3112 (14,400) A.R.C. Technical Report



ROYAL AIRC

### MINISTRY OF SUPPLY

AERONAUTICAL RESEARCH COUNCIL
REPORTS AND MEMORANDA

# A Frost-Point Hygrometer for Supersonic Wind Tunnels

(With Addendum)

By
D. Beastall and A. Winyard

C Crown copyright 1959

LONDON: HER MAJESTY'S STATIONERY OFFICE

1959

PRICE 3s. 6d. NET

# A Frost-Point Hygrometer for Supersonic Wind Tunnels

(With Addendum)

By

D. BEASTALL and A. WINYARD

COMMUNICATED BY THE DIRECTOR-GENERAL OF SCIENTIFIC RESEARCH (AIR),
MINISTRY OF SUPPLY

Reports and Memoranda No. 3112\*

July, 1951

Summary.—This report describes a frost-point hygrometer suitable for measuring the water-vapour content of the air in supersonic wind tunnels at any stagnation pressure within their present range of operation. It uses CO<sub>2</sub> as a coolant and is economical in construction and operation.

1. Introduction.—In order to avoid spurious readings in supersonic wind-tunnel tests it is essential to control the humidity of the air in the tunnel to some small value. This report describes a frost-point hygrometer suitable for measuring humidities down to at least 0.0005 lb of water/lb of air over the range of pressures at which variable-density wind tunnels operate. This requires a coolant capable of giving a minimum temperature of  $-40 \deg C$ , if the tunnel is to be operated at pressures down to  $2.5 \ \text{lb/sq}$  in. abs. with control to any intermediate temperature up to  $0 \deg C$  or above. The coolant used is carbon dioxide throttled from a high-pressure cylinder.

The main advantages of this hygrometer are:

- (a) Simplicity and cheapness of construction and operation
- (b) Speed with which a frost point can be obtained (especially desirable for intermittent tunnels)
- (c) Freedom from leaks (essential at pressures less than atmospheric)
- (d) Availability of coolant
- (e) Fine degree of control of temperature.

The hygrometer described here is in operation in the 9 in.  $\times$  9 in. Supersonic Wind Tunnel at the Royal Aircraft Establishment and has been used for a short time without any serious troubles.

A more elaborate form employing CO<sub>2</sub> coolant is in use in America and is described in Ref. 1.

<sup>\*</sup> Previously published as C.P. 106.



2. Description of Hygrometer.—Fig. 1 is a detailed drawing of the hygrometer. It consists essentially of an arrangement for cooling a metal disc over which is passed the air whose humidity is required. The formation of frost is noted visually; the temperature of the disc at which frost is deposited on it is measured by a thermo-couple, and the pressure of the air in the hygrometer is measured directly from a gauge. From this information the water content of the air can be derived using the chart in Fig. 2 reproduced from Ref. 2.

More specifically the construction and operation of the hygrometer is as follows:

It is connected (see Fig. 1) to a high-pressure carbon-dioxide cylinder; the CO<sub>2</sub> expands through a jet formed by a 6-in. length of 0.5mm outside diameter stainless-steel hypodermic tubing and impinges on the underside of a brass disc. Control of the temperature of the gas is by means of the pressure in the expansion chamber and this is regulated by the control valve (a needle valve). Fig. 3 shows the relationship between the pressure and temperature of the CO<sub>2</sub> when expanded from high pressure. It is essential during operation to keep the pressure in the expansion chamber above the triple point (approximately 5 Atm), in order to avoid choking the jet with solid CO<sub>2</sub>. This, therefore, is the lower limit of operation of the hygrometer and is — 70 deg F (— 56·7 deg C). The expansion-chamber pressure is recorded on an air-pressure gauge.

The tunnel air is directed onto the top of the disc from a  $\frac{1}{8}$ -in. diameter inlet pipe and is returned through the air outlet. To facilitate the observation of frost the top of the disc has a matt black finish. The disc is made of brass and is insulated from the body of the hygrometer by fibre washers to prevent excessive dissipation of the cooling effect of the  $CO_2$ . The temperature of the disc is measured by means of a copper-constantan thermo-couple junction (not shown in Fig. 1) soldered to the upper surface of the disc. The hole, through which the thermo-couple leads were taken, was made airtight by the use of Araldite. The lower cylindrical part of the hygrometer was originally intended as a filter to reduce the water content of the  $CO_2$  passing through the jet and thus reduce the possibility of icing up. It was found to be superfluous but has been left simply as a housing for the 6-in. length of 0.5-mm tubing comprising the jet.

Fig. 4 gives diagrammatically the lay-out of the installation on the tunnel. Pipes are connected to the settling chamber and return circuit of the tunnel and thus provide a pressure difference for passing the air through the hygrometer. The pressure gauge (in this case a mercury barometer capable of reading pressures in the range of tunnel-operating stagnation pressures) gives the air pressure in the hygrometer. The thermo-couple is connected to a millivoltmeter (scale 0 to 4 mV), the reference junction being immersed in transformer oil in a thermos flask. The thermo-couple was calibrated so that the disc temperature corresponding to the millivoltmeter reading was known.

Throughout the installation attention has been paid to the elimination of leaks especially on the inlet side of the air line to the hygrometer and the hygrometer itself has been subjected to a pressure test at 1,500 lb/sq in. to ensure that there is no leak of CO<sub>2</sub> to the upper surface of the disc.

3. Determination of the Water Vapour Content of a Sample of Air.—Allow the tunnel air to flow through the hygrometer by opening the taps in the inlet and outlet pipe lines. With the control valve closed, open the CO<sub>2</sub> cylinder valve; the expansion-chamber pressure will then be the cylinder pressure and no cooling of the disc will take place. Now, by opening the control valve slowly, the temperature of the disc can be lowered until frost appears; the reading of the millivoltmeter at which this occurs is then noted. The temperature of the reference junction and the air pressure in the hygrometer are then required. When the frost-point temperature has been calculated, the water vapour content can be derived using Fig. 2. The operating time for obtaining a frost point should not be more than half a minute. When closing the CO<sub>2</sub> cylinder valve, ensure that the control valve is set above 5 Atm to prevent blocking of the jet by solid CO<sub>2</sub>.

- 4. Further Development.—It should be realised that the arrangement described in this note is the outcome of a very short development programme and refinements may be made as experience is gained in its operation. Only the main dimensions are given in Fig. 1, since the successful operation seems to depend only on a few dimensions. One important point to note in the construction is that the CO<sub>2</sub> outlet should be large compared with the size of the jet to prevent solid CO<sub>2</sub> blocking the outlet. The only maintenance at present carried out is the periodical drying out of the hygrometer to prevent the accumulation of water deposited by the CO<sub>2</sub>. The upper and lower limits of cylinder pressure which cover the range of satisfactory operation of the hygrometer are not known at present. Perhaps the only major modification which seems desirable is the installation of an automatic method of recording frost to eliminate the human error in visual observation.
- 5. Acknowledgement.—The construction and installation of the hygrometer are due to Mr. N. Searle, who also offered helpful suggestions during its development.



#### ADDENDUM

Since writing this report, satisfactory operation of the hygrometer has been found to depend on the temperature of the CO<sub>2</sub> cylinder being above a certain value which is roughly 25 deg C. This temperature determines whether any moisture in the CO<sub>2</sub> will be deposited as ice in the jet of the hygrometer during operation, thus causing blockage.

If we treat the flow of the  $CO_2$  as a simple fluid-flow problem, the minimum temperature attainable in the jet can be expressed in terms of the temperature  $(T_0)$  of the  $CO_2$  in the cylinder. The jet is acting as the throat of a supersonic nozzle and the temperature in the jet will be that corresponding to sonic velocity of the  $CO_2$ . This minimum temperature, T, is given by the relation

Thus if  $T_0$  is above 45 deg C, T will be above 0 deg C and ice will not form in the jet. This, however, assumes no supersaturation of the water vapour in the  $CO_2$ . In practice it has been found that successful operation of the hygrometer is possible with cylinder temperatures of round about 25 deg C or above.



#### REFERENCES

No. Author Title, etc.
 1 T. L. Wheeler and H. O. McMahan OSRD 5151. May, 1945.
 2 R. V. Hensley ... ... Mollier diagrams for air saturated with water vapour at low temperatures. N.A.C.A. Tech. Note 1715. September, 1948.

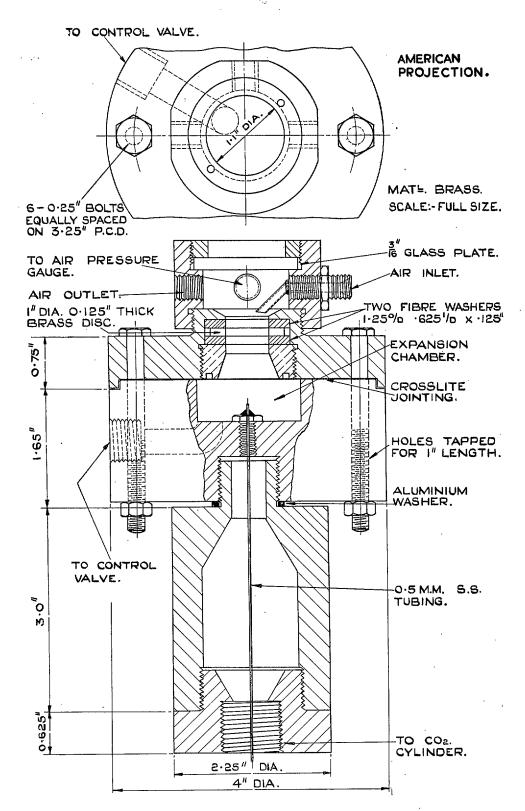


Fig. 1. Frost-point hygrometer.

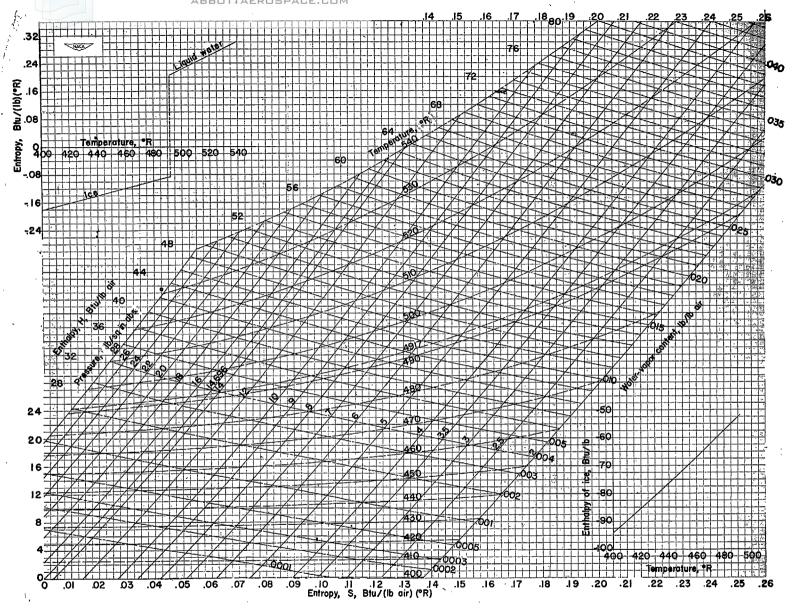


Fig. 2. Mollier diagram for air saturated with water vapour at vapour-solid equilibrium for sub-freezing temperatures.

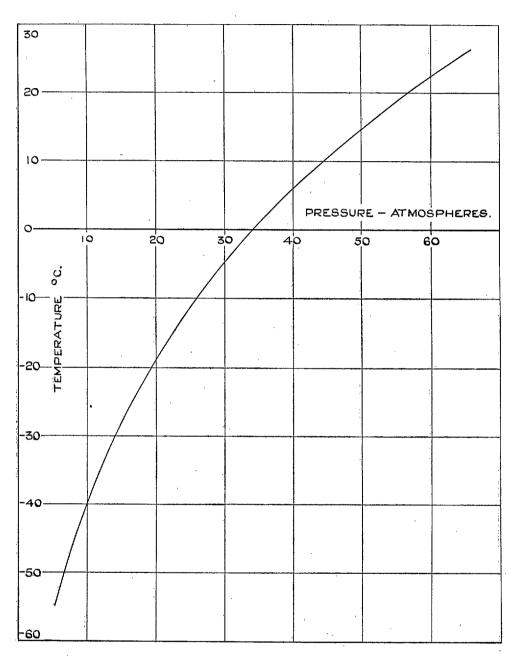


Fig. 3. Vapour pressure of CO<sub>2</sub>.

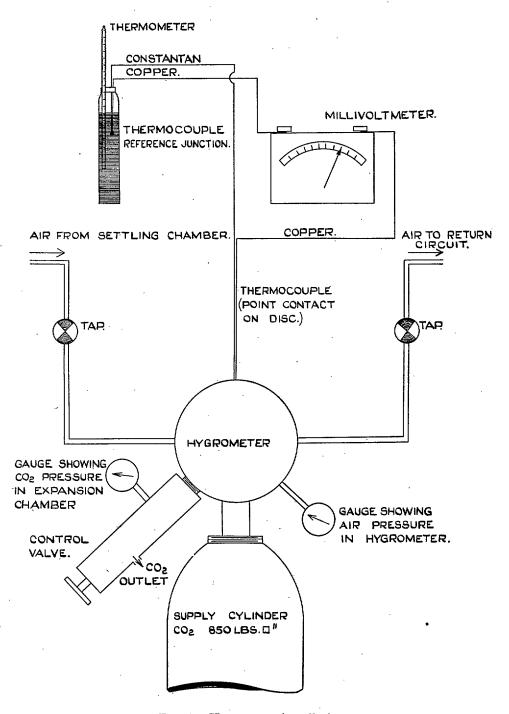


Fig. 4. Hygrometer installation.



# Publications of the Aeronautical Research Council

#### ANNUAL TECHNICAL REPORTS OF THE AERONAUTICAL RESEARCH COUNCIL (BOUND VOLUMES)

- 1939 Vol. I. Aerodynamics General, Performance, Airscrews, Engines. 50s. (52s.).
  - Vol. II. Stability and Control, Flutter and Vibration, Instruments, Structures, Seaplanes, etc. 63s. (65s.)
- 1940 Aero and Hydrodynamics, Aerofoils, Airscrews, Engines, Flutter, Icing, Stability and Control, Structures, and a miscellaneous section. 50s. (52s.)
- 1941 Aero and Hydrodynamics, Aerofoils, Airscrews, Engines, Flutter, Stability and Control, Structures. 63s. (65s.)
- 1942 Vol. I. Aero and Hydrodynamics, Aerofoils, Airscrews, Engines. 75s. (77s.)
- Vol. II. Noise, Parachutes, Stability and Control, Structures, Vibration, Wind Tunnels. 47s. 6d. (49s. 6d.)
- 1943 Vol. I. Aerodynamics, Aerofoils, Airscrews. 80s. (82s.)
  - Vol. II. Engines, Flutter, Materials, Parachutes, Performance, Stability and Control, Structures. 90s. (92s. 9d.)
- 1944 Vol. I. Aero and Hydrodynamics, Aerofoils, Aircraft, Airscrews, Controls. 84s. (86s. 6d.) Vol. II. Flutter and Vibration, Materials, Miscellaneous, Navigation, Parachutes, Performance, Plates and Panels, Stability, Structures, Test Equipment, Wind Tunnels. 84s. (86s. 6d.)
- 1945 Vol. I. Aero and Hydrodynamics, Aerofoils. 130s. (132s. 9d.)
  - Vol. II. Aircraft, Airscrews, Controls. 130s. (132s. 9d.)
  - Vol. III. Flutter and Vibration, Instruments, Miscellaneous, Parachutes, Plates and Panels, Propulsion. 130s. (132s. 6d.)
    Vol. IV. Stability, Structures, Wind Tunnels, Wind Tunnel Technique.
  - 130s. (132s. 6d.)

#### Annual Reports of the Aeronautical Research Council-

1937 2s. (2s. 2d.)

1938 is. 6d. (1s. 8d.)

1939-48 3s. (3s. 5d.)

#### Index to all Reports and Memoranda published in the Annual Technical Reports, and separately-

R. & M. 2600 2s. 6d. (2s. 10d.)

#### Author Index to all Reports and Memoranda of the Aeronautical Research Council—

1909-January, 1954

R. & M. No. 2570 15s. (15s. 8d.)

#### Indexes to the Technical Reports of the Aeronautical Research Council-

R. & M. No. 1850 1s. 3d. (1s. 5d.) December 1, 1936—June 30, 1939 July 1, 1939—June 30, 1945 R. & M. No. 1950 1s. (1s. 2d.) R. & M. No. 2050 1s. (1s. 2d.) R. & M. No. 2150 1s. 3d. (1s. 5d.) R. & M. No. 2250 1s. 3d. (1s. 5d.) July 1, 1945—June 30, 1946 July 1, 1946—December 31, 1946 January 1, 1947—June 30, 1947

#### Published Reports and Memoranda of the Aeronautical Research Council-

Between Nos. 2251-2349 R. & M. No. 2350 1s. 9d. (1s. 11d.) Between Nos. 2351-2449 R. & M. No. 2450 2s. (2s. 2d.) R. & M. No. 2550 2s. 6d. (2s. 10d.) R. & M. No. 2650 2s. 6d. (2s. 10d.) Between Nos. 2451-2549 Between Nos. 2551-2649 Between Nos. 2651-2749 R. & M. No. 2750 2s. 6d. (2s. 10d.)

Prices in brackets include postage

## HER MAJESTY'S STATIONERY OFFICE

York House, Kingsway, London W.C.2; 423 Oxford Street, London W.I; 13a Castle Street, Edinburgh 2; 39 King Street, Manchester 2; 2 Edmund Street, Birmingham 3; 109 St. Mary Street, Cardiff; Tower Lane, Bristol 1; 80 Chichester Street, Belfast, or through any bookseller.